

**Computational Stress and Deformation  
Analysis of Mammary Prosthesis**

**By Tavis L. Potter**

**Committee Chairman: J. W. Grant**

Engineering Mechanics

**(ABSTRACT)**

A linear and non-linear material model for the breast implants was developed through axial tension testing, while linear and non-linear breast tissue models were assumed based on smooth muscle. These material models were to develop axisymmetric finite element models to determine the stresses in the implant walls under tissue loading. The non-linear material models were used to more accurately model the complex nature of the implant stresses. After analysis it was found that the implants were under compressive loading which meant that local buckling in the implant might be possible.

For accurate stress prediction in the implant walls and to fully characterize implant buckling a more accurate non-linear breast tissue material model needs to be developed. Having this material model would allow for a full three-dimensional finite element model can be developed. With the development of a three-dimensional FEA model the implant buckling and implant stresses could be fully characterized. Ultimately allowing for accurate implant stress estimation and fatigue life calculation using the Palmgren-Miner rule, S-N curves, and an external load spectra.

## **Table Of Contents**

<b>Table Of Figures .....</b>	<b>v</b>
<b>List of Tables .....</b>	<b>vi</b>
<b>1. Introduction .....</b>	<b>1</b>
1.1. Literature Review.....	1
1.2. General Project Introduction.....	1
1.3. Materials introduction.....	3
1.4. FEA introduction .....	4
1.5. Fatigue introduction.....	7
1.6. Conclusions.....	8
<b>2. Material Properties .....</b>	<b>10</b>
2.1. General Material Properties .....	10
2.1.1 General Breast Tissue Formulation.....	11
2.1.2 General Implant Shell Material Formulation .....	13
<b>3. Finite Element Modeling .....</b>	<b>19</b>
3.1. General Finite Elements Information.....	19
3.1.1 Introduction .....	19
3.1.2 Initial Geometry .....	19
3.2. Implant Models and Results.....	28
3.2.1 Linear implant and tissue material models, with non-linear geometry, and no right side displacement constraint.....	28

3.2.2 Linear implant and tissue material models, with non-linear geometry and with a right side displacement constraint .....	35
3.2.3 Linear tissue and implant material models, with nonlinear geometry and inclusion of a fold .....	41
3.2.4 Nonlinear implant and tissue material models, with nonlinear geometry and no right side displacement constraint .....	48
3.2.5 Nonlinear implant and tissue material models, with nonlinear geometry and a right side displacement constraint.....	54
<b>4. Fatigue Introduction:.....</b>	<b>60</b>
4.1. Internal Stress Analysis: .....	60
4.1.1 Mechanical Testing .....	60
4.1.2 Loading Spectrum Estimation .....	62
4.1.3 Technical Data about Fatigue Life Estimation.....	62
4.1.4 S-N Curve.....	62
4.1.5 Loading Spectra and Cycle counting.....	63
4.1.6 Palmgren-Miner Rule .....	63
4.2. FEA Conclusions .....	64
<b>5. Conclusions .....</b>	<b>65</b>

## Table Of Figures

Figure 2.1.2.1-1 Load versus crosshead displacement .....	15
Figure 2.1.2.3-1 Elastic Modulus versus Crosshead speed.....	16
Figure 2.1.2.4-1 Dimensions and shape of the dogbone tension specimen. All dimensions are in mm.....	18
Figure 3.1.2-1: Tissue and implant model, including the mesh, for a non-fold model....	20
Figure 3.2.1.6-1: Von Mises stresses in the implant.....	31
Figure 3.2.1.6-2: Von Mises stresses in the tissue.....	32
Figure 3.2.1.6-3: Stress in the implant along the x direction.....	32
Figure 3.2.1.6-4: Stress in the tissue along the x direction.....	33
Figure 3.2.1.6-5: Shear stress in the tissue in the x-y plane.....	33
Figure 3.2.1.6-6: Stress in the implant along the y direction.....	34
Figure 3.2.1.6-7: Stress in the tissue along the y direction.....	34
Figure 3.2.2.6-1: Von Mises Stress in the implant .....	37
Figure 3.2.2.6-2: Von Mises stress in the tissue .....	38
Figure 3.2.2.6-3: Stress in the implant along the x direction.....	38
Figure 3.2.2.6-4: Stress in the tissue along the x direction.....	39
Figure 3.2.2.6-5: Shear stress in the tissue in the x-y plane.....	39
Figure 3.2.2.6-6: Stress in the implant along the y direction.....	40
Figure 3.2.2.6-7: Stress in the tissue along the y direction .....	40
Figure 3.2.3.1-1: Plot showing both the tissue and the implant, for the fold model.....	41
Figure 3.2.3.6-1: Von Mises stresses in the implant.....	44
Figure 3.2.3.6-2: Von Mises stresses in the tissue .....	44

Figure 3.2.3.6-3: Stress in the implant along the x direction.....	45
Figure 3.2.3.6-4: Stress in the tissue along the x direction .....	45
Figure 3.2.3.6-5: Shear stress in the tissue in the x-y plane.....	46
Figure 3.2.3.6-6: Stresses in the implant along y direction .....	46
Figure 3.2.3.6-7: Stress in the tissue along the y direction .....	47
Figure 3.2.4.6-1 Von Mises stresses in the implant.....	50
Figure 3.2.4.6-2 Von Mises stresses in the tissue.....	51
Figure 3.2.4.6-3 Stress in the implant along the x direction.....	51
Figure 3.2.4.6-4: Stress in the tissue along the x direction .....	52
Figure 3.2.4.6-5: Shear stress in the tissue in the x-y plane.....	52
Figure 3.2.4.6-6 Stress in the implant along the y direction .....	53
Figure 3.2.4.6-7: Stress in the tissue along the y direction .....	53
Figure 3.2.5.6-1: Von Mises in the implant stresses, note the bulge in the lower center of the implant, this is an example of an error in the model.....	56
Figure 3.2.5.6-2: Von Mises in the tissue, because of the contact modeling between the implant and the tissue the bulge also appears here .....	56
Figure 3.2.5.6-3: Stress in the implant along the x direction.....	57
Figure 3.2.5.6-4: Stress in the tissue along the x direction .....	57
Figure 3.2.5.6-5: Shear stress in the tissue in the x-y plane.....	58
Figure 3.2.5.6-6: Stress in the implant along the y direction .....	58
Figure 3.2.5.6-7: Stress in the tissue along the y direction .....	59

## **List of Tables**

Table 3-1 Table showing the Axisymmetric FEA models developed .....	21
---	----