(a) Finite-difference grid sensitivities $\partial y/\partial b_{10}$.
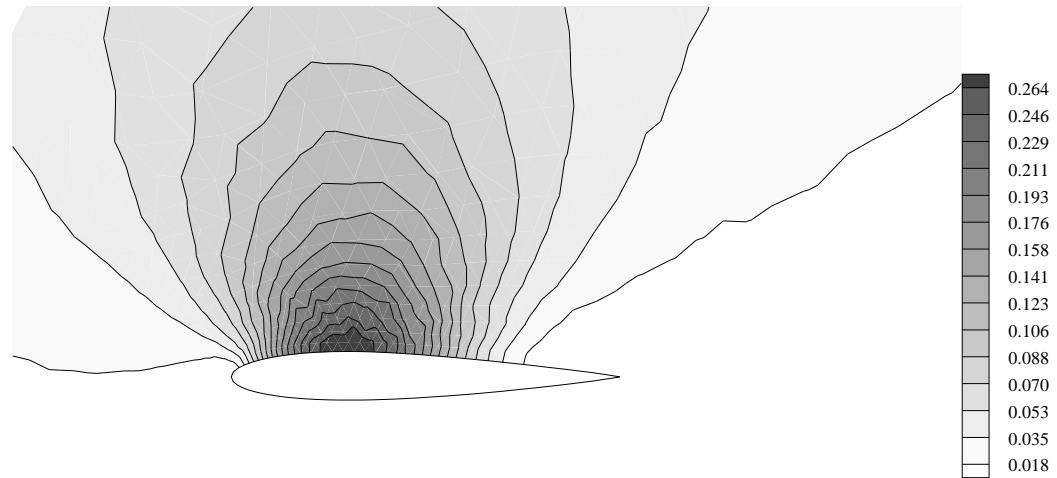
(b) ADIFOR generated grid sensitivities $\partial y/\partial b_{10}$.

Figure 6.2: Unstructured grid sensitivities for design variable 10 on the upper surface of the airfoil.