# A CAD/CAM Interface for Computer-Aided Design of Cams 

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# A CAD/CAM Interface for Computer-Aided Design of Cams 

by<br>Ashit R. Gandhi<br>Dr. Arvid Myklebust, Chairman<br>Mechanical Engineering<br>(ABSTRACT)

The purpose of this thesis is to provide a complete package for the design and three dimensional modeling display of cams.

The software produced as a part of this work will operate as a module of CADAM to produce cam designs and enter the resulting cam as a CAD model and produce the graphical display of the cam.

In addition to the introductory material, this thesis is divided into four sections. The section on the graphics packages used in this thesis includes a brief history and capabilities of each of the packages. The second section details the procedure to be adopted in order to design a cam. The next section details ANICAM, the program that has been developed to incorporate the design and display procedure. The fourth section of this thesis contains recommendations for further work in this area.

The theoretical work in this project is a combination of original derivations and applications of the theory in the design literature.

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## CHAPTER 1 INTRODUCTION

Today's fast changing technology has put tremendous pressure on corporations for the development of new systems and also to improve the current designs. This has turned out to be a complex, time consuming and often costly process. In the past, engineers would run into errors from even the most detailed blueprints. That meant redesigns which would cost several thousand dollars or more. The computer helped eliminate many errors in recent years, giving engineers more flexibility to experiment with a design before it was actually manufactured. Still much of that early Computer-Aided engineering was executed in $2-D$ on the computer screen, like sketching on an electronic piece of paper.

Today however the so called 3-D systems are available. The addition of depth, on 19-inch full-color screens has led to significant advances in engineering speed and accuracy. Using portable workstations, design engineers are able to work independently, as well as away from the office. This means that an engineer can check the designs in progress to avoid missteps as the design takes shape. It accelerates the communication between various engineers on a project and eliminates what could turn out to be costly errors. The reduced errors will also improve productivity. The $3-\mathrm{D}$ systems are able to store design information from the early stages of a project. Since engineers know how to retrieve it, they are free to spend more time on problem solving. Software packages developed can help engineers deduce where the moving parts will show early stress signs, which parts can be cut by the manufacturer from the same piece of material
(to be ground) and what illustrations might be best included in the user's manual.

The ability to model in $3-\mathrm{D}$ is significant since the designer no longer has to tie up time in manufacturing a model. This has been especially true when many design iterations are required to satisfy motion constraints. While many programs have been written to handle design and analysis, almost nothing can be found to link the analysis step to a 3-D modeler. Transferring data between these two steps has been very laborious. If a method had been available to do this automatically, a great savings in time and accuracy would have resulted. The purpose of this study is to create a program which will make it possible to tie these design steps together and present a method for creating 3-D animated models automatically.

This thesis introduces a program, ANICAM which has been created to animate the motion of cams in a 3-D format. It has been written in FORTRAN 77 computer language using the IBM4341 running the VM/CMS system and has been designed to serve as an interface between cam design and three existing programs: GKS, a graphics package, CADAM, and MOVIE.BYU which are packages for generating 3-D graphics and CAD models.

## CHAPTER 2 LITERATURE REVIEW.

The history of cams can be traced back several centuries. However it is very difficult to say where and when cams got started. In spite of being in use for a long time, it was only in the last four decades that major developments in the area of cam design have taken place.

While the kinematics of the cam-follower system have been thoroughly investigated by many including Rothbart(1), Jensen(2) and Chen(3), the absence of literature on the dynamic response has made it very difficult to validate analytic advances in this field. In recent years, investigations into the dynamics of the cam-follower systems have been performed by Johnson(4), Donkin and Clark(5) and others. However little experimental data, either in the form of system parameters or dynamic quantities, is available. Pisano(6) has gathered data on the description and recording of such measurements. The dynamics of cam-follower systems remains an active field of research and development.

The last two decades have seen a lot of work in the fields of design synthesis, design optimization, and computer-aided design. Many mathematical techniques have been used to find the optimal or best solution to the design problem. These consist of two phases: first, to generate alternative solutions and second, to identify the optimal solution among the acceptable ones based on certain criteria. Work on optimal design of the cam-follower systems has been done by Berzak(7) who optimally synthesized the system by the polydyne cam design method. Sermon and Liniecki(8) have used pressure angle and maximum stress constraints to
design minimum volume cams. Sandler(9) uses the spectral theory of random processes for the synthesis while Chen and Shah(10) have used the sequential random vector technique for the optimal design of geometric parameters of the system. Chew, Freudenstein and Longman(11) have suggested a model using the optimal-control theory.

The need to improve manufacturing productivity has generated large scale interest in computer-aided design and computer-aided manufacturing. In recent years, several computer programs have been written to perform basic design functions.

A number of programs for designing cam and follower systems are available. Most of these packages use the well known aspects of kinematic and static force analysis of cam mechanisms. COMMEND(12) is a program which considers almost every design aspect for the cam. An integrated program constisting of CAM, KAVM, and NCCAM(13) allows the design and graphical display of the cam-follower system. Other programs for the analysis and design of cam mechanisms include CAMPAK(14), DYCAM and CAMCHK (15). All the above programs allow two dimensional display of the physical characteristics and geometry of the cam-follower systems.

If computer graphics is to be exploited fully, then the design and analysis of the cam should be supported by high resolution, three dimensional shaded color images. Coupled with animation, this would give a realistic picture of the cam and follower. Three dimensional modeling packages like GEOMOD, CATIA, PADL-1 and 2, UNISOLIDS, $\operatorname{UNISCAD(16)~and~}$ recently CADAM have been developed by private corporations for commercial purposes. Another package, MOVIE.BYU, has been developed and distributed by the Brigham Young University. The availability of 3-D graphics pack-
ages has helped reduce the time required in tedious verifications of the practicality of the design.

While this review is by no means complete, it forms the backbone of this project which aims at developing a computer program (to be named ANICAM) for design and three dimensional display of cams.

## CHAPTER 3 GRAPHICS SUPPORT

### 3.1 GRAPHICAL KERNEL SYSTEM

Though standards in programming languages have existed for 25 years it was not until June 1982 that the Graphical Kernel System (GKS) was adopted by the International Organization for Standardization (ISO) as a draft International Standard. It was in the late 60's that some standard approaches began to appear in computer graphics. Had developments continued in this manner a standard would have been evolved much earlier. However in the late 60 's, interactive graphics required an expensive refresh display and a dedicated host computer. A total cost of $\$ 400,000$ was not uncommon and thus such systems were only available to a few. The advent of time sharing systems along with the emergence of storage tube display had a dramatic impact. Interactive graphics was now available to a larger number of users. The rapid changes in both hardware and patterns of usage left little time for evolving standardization in computer graphics. As widely differing devices were put into different facilities the need for a standard received more attention. Standards would help guard against the unproductive diversity of systems.

The process of working towards a common specification of a core graphics system began in August 1974, at an IFIP WG5. 2 meeting in Malmo, Sweden. In February 1977 a meeting of the programming language subcommittee ISO/TC97/WG5 was held in London. It was resolved that no existing graphics package could be a suitable candidate for a graphics standard.

It was decided that a working group of SC5 should be established to deal with the standardization process.

GKS was first reported in the first formal meeting of WG2 (the graphics working group) in Bologna in September 1978. It was decided that GKS be put forward to the parent body for registration as a Work Item with the aim of it becoming a Draft Proposal in a year. Technical meetings were held to resolve outstanding issues, incorporate changes into the GKS document and present it to the parent body. GKS was accepted as a standard in June 1982. GKS was revised and is now Draft International Standard (DIS) 7942(10).

### 3.1.1 CAPABILITIES OF GKS

GKS is a subroutine package that provides functions for managing graphical output and input. It is a general-purpose graphics package that provides facilities designed to address a wide range of applications. An application program uses it as a standard interface to graphic devices. The package is a large one with well over a hundred functions defined. Although its size gives GKS the appearance of complexity, the concepts embodied in the package are quite simple. The principal concepts of GKS are enumerated in the following list:

- Separate functions are provided for input and output. There is some interplay between the functions.
- Graphical primitives for output go either directly to the display or into a segment. Segments are used to manage interactive changes to a display.
- Primitive attributes are set to control the appearance of each graphical output primitive.
- Segment attributes affect all the primitives in a segment.
- The regeneration of the display from a set of segments can be controlled explicitly. GKS may also provide implicit or immediate display updating.
- Graphical input is provided by a set of logical devices, which are not really "devices", but interactive techniques for obtaining a particular kind of information.
- Inquiry functions allow the application program to obtain a great deal of information about the output and input devices available to GKS. The program can use this information to tailor its operation to the capabilities of the device.

GKS provides functions for 2-D graphics only; extensions to GKS now being designed for 3-D graphics. A complementary standard for 3-D grahics and animation, PHIGS, is under development. At present, 3-D objects can be displayed if the application program projects the $3-D$ information to $2-D$ before calling GKS. The most common approach is to make a perspective drawing of the object. Perspective, however, is not the only mechanism for communicating depth information to the viewer. Images that show solid objects with simulated lighting, shadows, texture and color can achieve
startling realism. Unlike making perspective drawings, generating these pictures, require a great deal of computations.

The GKS functions are a set of tools for making a display device present the picture or for implementing an interactive graphical dialog the user has designed. While understanding the functions is essential to undertake a programming project, a feeling for the strategy used to design an interactive program is also essential for the tactics that get the most out of GKS.

### 3.2 CADAM

The CADAM (Computer-graphics Augmented Design and Manufacturing) system is an integrated set of computer programs with a common data base that can be used to prepare mechanical drawings on a computer terminal, similar to using a drawing board. This system can be used for design and manufacturing tasks in a wide variety of industries. CADAM programs were first developed and marketed in the mid 60's by the Lockheed Aircraft Corporation. Today, the CADAM system is one of the most widely used packages throughout the world. Recently a new version of the CADAM system was released. The system now has 3-D capabilities, allowing the user to create wire frame models.

### 3.2.1 CAPABILITIES OF CADAM

The CADAM software is a set of interrelated computer programs that allow the user to perform a variety of design or NC (Numerical Control)
tasks. The system uses a computer and various peripheral computer devices to accomplish this task. The capabilities of the CADAM system are so widespread and versatile that it is impossible to dictate a mode of use. Detailed drawings from any model can be picked in a matter of seconds, making the system well suited for layout design. By combining user design skills with the power of the computer, it is possible to solve a variety of engineering problems much faster and with greater accuracy. The CADAM system does much of the drafting work, freeing the user from the repetitious job of drawing and redrawing the same objects over and over again. An interesting and extremely useful feature is the Geometry Interface module which allows the user to access and modify the CADAM data base. This can be accomplished by the following methods:

CREATION of a collection of CADAM elements (a CADAM model) and adding it to the collection of CADAM models.

RETRIEVAL from a collection of CADAM models for use in the user application program.

Each geometry interface function is separate and distinct. There are 4 geometry interface programs. They are listed below with a brief description of the modules used in this thesis work.

LOFT takes curve and surface shape definitions and assembles them into a CADAM model which can be stored to the CADAM data base.

CADET is a collection of CALL statements which calls passive subroutines that are designed to receive disassembled CADAM elements.

CADCD provides a collection of subroutines which are driven by CALL statements to produce CADAM elements which can be added as a CADAM model to a drawing file. Using CADCD it is possible to put new drawings in the data base, make additions to existing drawings or to add details in a batch mode. All the user needs to do is to write an application program using the subroutines and link it together with the CADCD module.

CADMACGM is a collection of subroutines that are driven by CALL statements to produce CADAM elements. In fact it is an enhanced version of the CADCD module and contains all CADCD subroutine calls except data management.

All parameters in the calling sequence in CADCD and CADMACGM are either REAL*4 or INTEGER*4 and follow default FORTRAN naming conventions. It must be noted that CADCD and CADMACGM use blank common. Hence the user must be extremely careful while passing any information to these module subroutines. Since the blank common will override information passed, the user must not make use of blank common in his application program. An useful variation of using the CADCD subroutine is Macro Geometry.

Macro Geometry allows the user to build load modules from the CADCD subroutines to be executed from the CADAM scope. This capability is created by linking the CADMACGM module (instead of CADCD) with the user developed application program. The program is accessed via the /MACRO GEO/ menu option of the function key DETAIL for the interactive CADAM system. This displays a list of available application programs and the
required parameters whose program names, variable lists and/or menus were previously defined by an application programmer through the use of the CADPARMC assembler routine.

### 3.3 MOVIE.BYU

In 1966 the Computer Science department at the University of Utah, under the direction of David Evans, began to develop a hardware and software capability in graphics. The main thrust of the research work was in the area of continuous tone image generation. By 1968 Hank Christiansen had joined with Evans to explore engineering applications for shaded pictures. This led to the development of MOVIE.BYU, a general purpose computer graphics software system. The MOVIE.BYU system requires a time sharing digital computer with a word length of at least 32 bits. The system has been distributed by Brigham Young University since 1976.

The original display programs of MOVIE. BYU were written at the University of Utah and used to produce raster driven displays with the option of a visible surface processor to solve the hidden surface problem in hardware. This problem was solved with the availability of Watkin's algorithm (24) which was optionally called by the display program if the hardware processor was down. The development of new editions of the program led to significant improvement in the line drawing capabilities of the routines. This system was distributed under the name MOVIE.NUC. Research by Mike Stephenson and Christiansen at Utah produced an expanded capability for MOVIE as well as refinement of data editing routines, character generators, and the capability to process 3-D finite element
models. These developments were incorporated into the MOVIE.NUC program thus producing the 1976 edition of MOVIE.BYU. Since then revisions have been made each year.

### 3.3.1 CAPABILITIES OF MOVIE

The elements of the MOVIE system ( DISPLAY, UTILITY, SECTION, TITLE, MOSAIC and COMPOSE ) are FORTRAN programs for display and manipulation of data representing mathematical, topological and architectural models whose geometries may be described in terms of polygonal elements or contour line definitions. These six programs work in harmony to provide displays of the data in line drawing or continuous tone image format, to clip and cap 3-D systems to expose internal surfaces, to modify geometry, displacement, and/or scalar function files by way of correction, appendage, or symmetry operation, to generate new models or title representations, to convert complex contour line definitions into polygonal element mosaics, or to view multiple models simultaneously. A brief description of each program will help the user to understand their interrelationships.

DISPLAY is the heart of the system. It is the display program for the polygonal element models. The executable program consists of a minimum of six source files compiled and loaded together. The program allows the model to be translated in any direction and rotation about any of the global or local cartesian axes using rigid body commands. The capability to select color for the background and individual parts and the ability
to shade give the system an excellent capacity to produce a smooth surface simulation. Animated sequences involving the scene manipulation commands allow the specification of harmonic oscillations and incremental or smooth animation of the rigid body motion. The program also allows casting of shadows from multiple light sources, transparency, fog simulation and anti-aliasing.

UTILITY is the data generation and editing program which allows the user to produce and/or edit models of 2-D or 3-D polygonal systems. The program allows the generation of 2 and $3-D$ finite element models in shell or solid element format. The user may also use UTILITY to read data files for the purpose of modification, appendage to other files or to subject the model to symmetry operations.

SECTION is a special purpose program to modify solid data representations so that they are compatible with the display program. The algorithm also allows the repeated dividing of a solid model along a arbitrary set of planes and user defined curves in space, generation of elements on the cut surface and the deletion of all elements and corresponding coordinates, displacements and scaler functions that are interior to the model.

TITLE is a program to generate 2 and 3-D characters whose data format is the same as that used by other programs. The user enters a line of text which is converted into characters composed of polygons according to specifications given by the user.

MOSAIC is a program for processing complex contours to produce element mosaics which are suitable for line drawing and continuous tone display. MOSAIC includes node thinning and selective reduction of triangular pairs to quadrilaterals.

COMPOSE is a program which, in combination with a "Record" command in DISPLAY, allows the user to create multiple image line drawings. Several line drawings are saved on disk as named ASCII files before the user runs the COMPOSE program which retrieves the files and builds up multiple image displays in an automatic or manual mode.

## CHAPTER 4 DESIGN CONSIDERATION

### 4.1 INTRODUCTION

It is almost impossible to produce a cam which can meet all its functional requirements. Since present day requirements usually dictate that components move in a prescribed, exact path, it is important to have a cam which satisfies the displacements exactly. In most of the cases the designer is confronted with a set of requirements which must be satisfied. Generally, a cam must be optimized to meet most of its requirements. Thus the cam produced by the optimization technique would be the best one for a given set of conditions. A cam may be designed in two ways:

1. By assuming the required motion for the follower and designing the cam to give this motion.
2. By assuming the shape of the cam and determining what characteristics of displacement, velocity and acceleration this contour will give.

The first method is a good example of synthesis. In fact, designing a cam mechanism from the desired motion is an application of synthesis that can be solved every time. Hence, in any cam mechanism the first decision always relates to motion specifications. In order to select a simple, optimum cam profile it is necessary to develop a complete timing diagram showing the displacement of the mechanism and its proper spacing and place in the time cycle. A good cam profile is often one that has the lowest
maximum acceleration. If there are other considerations to be met, a compromise must be made with other optimizing parameters. On the other hand, if two or more mechanisms must be synchronized and have their motions closely coordinated, it may be necessary to use a less than optimum cam profile for a part of the total displacement. This brings about the need for careful boundary matching and the blending together of composite profiles.

Very little is available concerning methods of selecting the most suitable profile for a given application and of accurately producing the profile in cam manufacture. This difficulty is eliminated by making the cam symmetrical and by using shapes for cam contours that can be generated easily. Besides motion considerations, a primary concern of the designer must be the type and magnitude of the load between the cam and follower. For low speeds and low masses, the inertia load is low and will cause no appreciable deflection of components. Hence, almost any cam curve can be used. At high speeds (or masses), however, the inertia load increases and the cam mechanism becomes prone to deflection and creates vibrations which will produce vibratory stresses in the follower system. Now the cam profile becomes of utmost importance. Figure 1 displays the flow chart for the basic approach for cam design.

### 4.2 CAM DISPLACEMENT CURVES

There are infinitely many follower motions that can be used for the rise and return of the follower. There are a number of so called "basic curves" like Constant Velocity or Straight Line, Constant Acceleration


Figure 1. Flow chart for basic approach for Cam design
or Parabolic and trigonometric curves like Simple Harmonic, Cycloidal and Elliptical curves. These basic curves may be inadequate on some occasions, especially in high speed application. An alternate, versatile manner of motion specification is the use of polynomial function. Solutions for the basic curves and low-order power polynomials can be easily obtained. However, as the conditions of control become complex, the order of the polynomial increases and the designer runs into numerical difficulties. In an effort to incorporate the better features of the basic curves into one optimum curve, designers have tried many combinations of basic curves. Composite curves like Trapezoidal and Fourier Series curves have been formed to produce minimum acceleration of (and hence, minimum dynamic load on) the follower. Certain fundamental conditions must be satisfied while combining curves. It should be noted that any combination of the basic curves may be utilized to fulfill the requirements of the fundamental conditions. The following conditions must be satisfied:

1. The sum of the displacements during each of the various motions should be equal to the total stroke.
2. The sum of the angles turned through by the cam during each of the various motions should be equal to the total cam angle.
3. The velocities of all curves at the junction should be equal.
4. High-speed action requires that acceleration of all curves at the junction be equal.

Based on the fundamental conditions of combining cam curves, a customized cam can be developed with a predesigned building block of cam curve seg-
ments. This thesis considers 22 different motion curves. A brief description is provided for each type of curve.

### 4.2.1 SIMPLE HARMONIC MOTION

This motion is characterized by having its acceleration proportional and in an opposite direction to its displacement. The usefulness of these curves is further expanded by using the different forms of SHM as curve segments to form a single contour. If $L$ is the total lift, $\beta$ the active cam angle and $\theta$ the angle of rotation of the cam then the motion characteristics of the curve segments are represented by equations 1 through 18.

H-1 Harmonic: As seen from Fig. 2 this is a lifting curve which has a constant velocity at the end of the stroke.

$$
\begin{align*}
& s=L\left\{1-\cos \left(\frac{\pi \theta}{2 \beta}\right)\right\}  \tag{1}\\
& v=\frac{\pi L}{2 \beta} \sin \left(\frac{\pi \theta}{2 \beta}\right)  \tag{2}\\
& a=\frac{\pi^{2} L}{4 \beta^{2}} \cos \left(\frac{\pi \theta}{2 \beta}\right) \tag{3}
\end{align*}
$$





Figure 2. Motion Characteristics for H-1 curve


H-2 Harmonic: In the case of the H-2 lift curve the velocity goes to zero at the end of the stroke. However the acceleration at this point is not zero. The characteristics are as shown in Fig. 3.

$$
\begin{equation*}
s=L \sin \left(\frac{\pi \theta}{2 \beta}\right) \tag{4}
\end{equation*}
$$

$$
\begin{equation*}
v=\frac{\pi L}{2 \beta} \cos \left(\frac{\pi \theta}{2 \beta}\right) \tag{5}
\end{equation*}
$$

$$
\begin{equation*}
a=-\frac{\pi^{2} L}{4 \beta^{2}} \sin \left(\frac{\pi \theta}{2 \beta}\right) \tag{6}
\end{equation*}
$$

H-3 Harmonic: H-3 is a descending curve and has properties symmetric to that of H-2. Hence it would be an ideal curve for combination with that curve. Figure 4 shows the motion characteristics for this curve segment.

$$
\begin{align*}
& s=L \cos \left(\frac{\pi \theta}{2 \beta}\right)  \tag{7}\\
& v=-\frac{\pi L}{2 \beta} \sin \left(\frac{\pi \theta}{2 \beta}\right)  \tag{8}\\
& a=-\frac{\pi^{2} L}{4 \beta^{2}} \cos \left(\frac{\pi \theta}{2 \beta}\right) \tag{9}
\end{align*}
$$

H-4 Harmonic: While this is a descending curve too, its characteristics, as seen from Fig. 5, are different from that of H-3.


acceleration
Figure 4. Motion Characteristics for H-3 curve




acceleration
Figure 5. Motion Characteristics for H-4 curve

$$
\begin{align*}
& s=L\left(1-\sin \left(\frac{\pi \theta}{2 \beta}\right)\right)  \tag{10}\\
& v=-\frac{\pi L}{2 \beta} \cos \left(\frac{\pi \theta}{2 \beta}\right)  \tag{11}\\
& a=\frac{\pi^{2} L}{4 \beta^{2}} \sin \left(\frac{\pi \theta}{2 \beta}\right) \tag{12}
\end{align*}
$$

H-5 Harmonic: H-5 curve is the most frequently used harmonic curve. This is because it produces a smooth transition point for the combining curve. H-5 characteristics are shown in Fig. 6.

$$
\begin{align*}
& s=\frac{L}{2}\left(1-\cos \left(\pi \frac{\theta}{\beta}\right)\right)  \tag{13}\\
& v=\frac{\pi L}{2 \beta} \sin \left(\pi \frac{\theta}{\beta}\right)  \tag{14}\\
& a=\frac{\pi^{2} L}{2 \beta^{2}} \cos \left(\pi \frac{\theta}{\beta}\right) \tag{15}
\end{align*}
$$

H-6 Harmonic: The $\mathrm{H}-6$ curve is often used in conjunction with the $\mathrm{H}-5$ curve and has properties symmetric to it. The motion characteristics of H-6 are as shown in Fig. 7.

$$
\begin{equation*}
s=\frac{L}{2}\left(1+\cos \left(\pi \frac{\theta}{\beta}\right)\right) \tag{16}
\end{equation*}
$$






Figure 6. Motion Characteristics for H-5 curve




Figure 7. Motion Characteristics for H-6 curve

$$
\begin{align*}
& v=-\frac{\pi L}{2 \beta} \sin \left(\pi \frac{\theta}{\beta}\right)  \tag{17}\\
& a=-\frac{\pi^{2} L}{2 \beta^{2}} \cos \left(\pi \frac{\theta}{\beta}\right) \tag{18}
\end{align*}
$$

### 4.2.2 CYCLOIDAL MOTION

This motion is obtained by rolling a circle on a straight line and generating the locus of a fixed point on the circumference of the circle. From the generating principle it is easy to write the motion equations for these curve segments (Equations 19 to 36).

C-1 Cycloidal: The motion characteristics of this curve are shown in Fig. 8. As seen from the figure, this curve has characteristics similar to the $\mathrm{H}-1$ curve.

$$
\begin{align*}
& \mathbf{s}=L\left(\frac{\theta}{\beta}-\frac{1}{\pi} \sin \left(\pi \frac{\theta}{\beta}\right)\right)  \tag{19}\\
& v=\frac{L}{\beta}\left(1-\cos \left(\pi \frac{\theta}{\beta}\right)\right) \tag{20}
\end{align*}
$$

$$
\begin{equation*}
a=\frac{\pi L}{\beta^{2}} \sin \left(\pi \frac{\theta}{\beta}\right) \tag{21}
\end{equation*}
$$





Figure 8. Motion Characteristics for C-1 curve


Figure 9. Motion Characteristics for C-2 curve



C-2 Cycloidal: Figure 9 shows the motion characteristics of the C-2 curve. Its velocity and acceleration are symmetric to that of the C-1 curve.

$$
\begin{align*}
& s=L\left(\frac{\theta}{\beta}+\frac{1}{\pi} \sin \left(\pi \frac{\theta}{\beta}\right)\right.  \tag{22}\\
& v=\frac{L}{\beta}\left(1+\cos \left(\pi \frac{\theta}{\beta}\right)\right.  \tag{23}\\
& a=-\frac{\pi L}{\beta^{2}} \sin \left(\pi \frac{\theta}{\beta}\right) \tag{24}
\end{align*}
$$

C-3 Cycloidal: The motion characteristics of this curve are shown in Fig. 10.

$$
\begin{gather*}
s=L\left(1-\frac{\theta}{\beta}+\frac{1}{\pi} \sin \left(\pi \frac{\theta}{\beta}\right)\right)  \tag{25}\\
v=-\frac{L}{\beta}\left(1-\cos \left(\pi \frac{\theta}{\beta}\right)\right)  \tag{26}\\
a=-\frac{\pi L}{\beta^{2}} \sin \left(\pi \frac{\theta}{\beta}\right) \tag{27}
\end{gather*}
$$

C-4 Cycloidal: As Fig. 11 shows, the C-4 curve has charecteristics which are symmetrically opposite to that of C-1 curve.



Figure 10. Motion Characteristics for C-3 curve


$$
\begin{gather*}
s=L\left(1-\frac{\theta}{\beta}-\frac{1}{\pi} \sin \left(\pi \frac{\theta}{\beta}\right)\right)  \tag{28}\\
v=-\frac{L}{\beta}\left(1+\cos \left(\pi \frac{\theta}{\beta}\right)\right)  \tag{29}\\
a=\frac{\pi L}{\beta^{2}} \sin \left(\pi \frac{\theta}{\beta}\right) \tag{30}
\end{gather*}
$$

C-5 Cycloidal: The C-5 curve is used very often in conjunction with the C-6 curve. The combination of these curves gives continuous velocity and acceleration curves. Figure 12 shows the characteristics of the C-5 curve.

$$
\begin{align*}
& s=L\left(\frac{\theta}{\beta}-\frac{1}{2 \pi} \sin \left(2 \pi \frac{\theta}{\beta}\right)\right)  \tag{31}\\
& v=\frac{L}{\beta}\left(1-\cos \left(2 \pi \frac{\theta}{\beta}\right)\right)  \tag{32}\\
& a=\frac{2 \pi L}{\beta^{2}} \sin \left(2 \pi \frac{\theta}{\beta}\right) \tag{33}
\end{align*}
$$

C-6 Cycloidal: Figure 13 shows the characteristics of this curve.

$$
\begin{equation*}
s=L\left(1-\frac{\theta}{\beta}+\frac{1}{2 \pi} \sin \left(2 \pi \frac{\theta}{\beta}\right)\right) \tag{34}
\end{equation*}
$$




$$
\begin{equation*}
v=-\frac{L}{\beta}\left(1-\cos \left(2 \pi \frac{\theta}{\beta}\right)\right) \tag{35}
\end{equation*}
$$

$$
\begin{equation*}
a=-\frac{2 \pi L}{\beta^{2}} \sin \left(2 \pi \frac{\theta}{\beta}\right) \tag{36}
\end{equation*}
$$

As there are no sudden changes in acceleration at the dwell ends, the cycloidal cam is has the smoothest motion among all basic curves, and is therefore suitable for high-speed applications.

### 4.2.3 CONSTANT VELOCITY

This curve has the simplest form of the polynomial (or a straightline) equation. If $s$ is the follower displacement at a cam angle $\theta$ and $C_{0}$ and $C_{1}$ are constants then this form can be represented as,

$$
\begin{equation*}
s(\theta)=C_{0}+C_{1} \theta \tag{37}
\end{equation*}
$$

Applying the end conditions to equation 37 we get,

$$
\begin{equation*}
s=\frac{h}{\beta} \theta \tag{38}
\end{equation*}
$$

where, $h$ is the total rise during cam rotation $\beta$. The equations for velocity, acceleration and jerk follow from this equation.

$$
\begin{equation*}
v=\frac{h}{\beta} \tag{39}
\end{equation*}
$$

$$
\begin{equation*}
a=0 \tag{40}
\end{equation*}
$$

Motion characteristics for constant velocity are shown in Fig. 14.

### 4.2.4 CONSTANT ACCELERATION

This curve, also known as parabolic or gravity curve, has constant positive and negative accelerations. The curve may have different displacement functions on either side of the point of inflection. If the inflection point occurs midway through the stroke, then, the motion equations are

For $0 \leq \theta \leq \beta / 2$

$$
\begin{align*}
& s=\frac{2 h}{\beta^{2}} \theta^{2}  \tag{41}\\
& v=\frac{4 h}{\beta^{2}} \theta  \tag{42}\\
& a=\frac{4 h}{\beta^{2}} \tag{43}
\end{align*}
$$

For $\beta / 2 \leq \theta \leq \beta$

$$
\begin{equation*}
s=h-\frac{2 h}{\beta^{2}}(\beta-\theta)^{2} \tag{44}
\end{equation*}
$$


nolluag'aoju


$\forall$ LaHL

$$
\begin{equation*}
v=\frac{2 h}{\beta}(\beta-\theta) \tag{45}
\end{equation*}
$$

$$
\begin{equation*}
a=-\frac{2 h}{\beta^{2}} \tag{46}
\end{equation*}
$$

In both cases, the jerk is zero, except where changes in acceleration occur, in which case jerk is infinite. Figure 15 shows the motion characteristics for this type of motion.

### 4.2.5 POLYNOMIAL CURVES

As the boundary conditions of the motion increase it becomes necessary to use higher degree polynomial equations. Equation 47 shows an nth degree polynomial equation which can be used to satisfy any number of boundary conditions.

$$
\begin{equation*}
s=C_{0}+C_{1} \theta+C_{2} \theta^{2}+\ldots .+C_{n} \theta^{n} \tag{47}
\end{equation*}
$$

This technique can produce smooth and aesthetically acceptable profiles that fulfill design requirements, although the mathematics involved may sometimes be quite cumbersome. For the sake of convenience, we shall normalize both, the total rise ( $h$ ) and the active cam angle ( $\beta$ ) i.e. $h$ $=1$ and $\beta=1$. The $2-3,3-4-5$, and $4-5-6-7$ polynomial curves fall under this category. Figures 16,17 , and 18 represent the normalized motion characteristics of the three polynomials respectively. The normalized motion equations are represented by equations 48 through 56 .

วлธทว








DISPLACEMENT

Figure 17. Motion Characteristics for normalized 3-4-5 Polynomial


acceleration
Figure 18. Motion Characteristics for normalized 4-5-6-7 Polynomial
 Curve

## 2-3 Polynomial

$$
\begin{align*}
& s=3 \theta^{2}-2 \theta^{3}  \tag{48}\\
& v=6 \theta-6 \theta^{2}  \tag{49}\\
& a=6-12 \theta \tag{50}
\end{align*}
$$

3-4-5 Polynomial

$$
\begin{align*}
& s=10 \theta^{3}-15 \theta^{4}+6 \theta^{5}  \tag{51}\\
& v=30 \theta^{2}-60 \theta^{3}+30 \theta^{4}  \tag{52}\\
& a=60 \theta-180 \theta^{2}+120 \theta^{3} \tag{53}
\end{align*}
$$

4-5-6-7 Polynomial

$$
\begin{gather*}
s=35 \theta^{4}-84 \theta^{5}+70 \theta^{6}-20 \theta^{7}  \tag{54}\\
v=140 \theta^{3}-420 \theta^{4}+420 \theta^{5}-140 \theta^{6} \tag{55}
\end{gather*}
$$

$$
\begin{equation*}
a=420 \theta^{2}-1680 \theta^{3}+2100 \theta^{4}-840 \theta^{5} \tag{56}
\end{equation*}
$$

### 4.2.6 FOURIER SERIES CURVES

There are certain disadvantages associated with the parabolic and cycloidal curves. They are likely to induce resonance vibrations or high acceleration. Any general purpose cam can be optimized with a curveintermediate between the parabolic and the cycloidal, but with the fewest and the lowest possible harmonics. Based on the above several harmonic series have been proposed. The kinematic equations of three are given below:

Gutman 1-3 Harmonic(90): This curve is obtained from the Fourier series expansion of the parabolic curve by retaining the first two terms of the series. Using standard notations, the motion equations can be represented as follows:

$$
\begin{align*}
& s=h\left(\frac{\theta}{\beta}-\frac{15}{32 \pi} \sin 2 \pi \frac{\theta}{\beta}-\frac{1}{96 \pi} \sin 6 \pi \frac{\theta}{\beta}\right)  \tag{57}\\
& v=\frac{h}{\beta}\left(1-\frac{15}{16 \pi} \cos 2 \pi \frac{\theta}{\beta}-\frac{1}{16 \pi} \cos 6 \pi \frac{\theta}{\beta}\right) \tag{58}
\end{align*}
$$

$$
\begin{equation*}
a=\frac{h \pi}{8 \beta^{2}}\left(15 \sin 2 \pi \frac{\theta}{\beta}+3 \sin 6 \pi \frac{\theta}{\beta}\right) \tag{59}
\end{equation*}
$$

Figure 19 shows the motion characteristics for this curve.

Freudenstein 1-3 Harmonic(84): This curve has been developed to minimize the dynamic acceleration factor. The equations are

$$
\begin{gather*}
s=h\left(\frac{\theta}{\beta}-\frac{h}{2 \pi} \frac{27}{28} \sin 2 \pi \frac{\theta}{\beta}+\frac{1}{84} \sin 6 \pi \frac{\theta}{\beta}\right)  \tag{60}\\
v=\frac{h}{\beta}\left(1-\frac{27}{28} \cos 2 \pi \frac{\theta}{\beta}-\frac{1}{28} \cos 6 \pi \frac{\theta}{\beta}\right)  \tag{61}\\
a=\frac{2 \pi h}{\beta^{2}}\left(\frac{27}{28} \sin 2 \pi \frac{\theta}{\beta}+\frac{3}{28} \sin 6 \pi \frac{\theta}{\beta}\right) \tag{62}
\end{gather*}
$$

The motion functions are shown in Fig. 20.

Freudenstein 1-3-5 Harmonic(84): Corresponding equations for this curve are

$$
\begin{align*}
& s=\frac{h \theta}{\beta}-\frac{h m}{2 \pi}\left(\sin 2 \pi \frac{\theta}{\beta}+\frac{1}{54} \sin 6 \pi \frac{\theta}{\beta}+\frac{1}{1250} \sin 10 \pi \frac{\theta}{\beta}\right)  \tag{63}\\
& v=\frac{h}{\beta}\left(1-m\left(\cos 2 \pi \frac{\theta}{\beta}+\frac{1}{18} \cos 6 \pi_{\beta}^{\theta}+\frac{1}{250} \cos 10 \pi \frac{\theta}{\beta}\right)\right)  \tag{64}\\
& a=\frac{2 \pi h}{\beta^{2}}\left(m\left(\sin 2 \pi \frac{\theta}{\beta}+\frac{1}{6} \sin 6 \pi \frac{\theta}{\beta}+\frac{1}{50} \sin 10 \pi \frac{\theta}{\beta}\right)\right) \tag{65}
\end{align*}
$$




Figure 19. Motion Characteristics for Gutman 1-3 Harmonic Curve




where,
$m=\frac{1125}{1192}$

The motion characteristics are shown in Fig. 21.

### 4.2.7 MODIFIED CAM CURVES

Modified curves were developed to use the better features of the basic curves in designing cams with the lowest maximum acceleration and having bounded jerk. Two such curves are modified sine and modified trapezoidal curves. The characteristics of these curves are shown in Fig. 22 and Fig. 23.

Modified Sine: This composite curve is a combination of cycloidal and harmonic curves. It consists of 3 sections, each of which is governed by different displacement functions. Application of the boundary conditions, matching of velocities and accelerations, and sustituting $\gamma$ for $\theta / \beta$ lead to equations 66 to 74 .

For $0 \leq \gamma \leq 1 / 8$

$$
\begin{gather*}
s=h(0.43989 \gamma-0.035014 \sin (4 \pi \gamma))  \tag{66}\\
v=0.43989 \frac{h}{\beta}(1-\cos (4 \pi \gamma)) \tag{67}
\end{gather*}
$$








Figure 22. Motion Characteristics for Modified Sine Curve



displacenen

Figure 23. Notion Characteristics for Modified Trapezoidal Curve

$$
\begin{equation*}
a=5.52794 \frac{h}{\beta^{2}} \sin (4 \pi \gamma) \tag{68}
\end{equation*}
$$

For $1 / 8 \leq \gamma \leq 7 / 8$

$$
\begin{gather*}
s=h\left(0.28005+0.43989 \gamma-0.315055 \cos \left(\frac{4 \pi}{3} \gamma-\frac{\pi}{6}\right)\right)  \tag{69}\\
v=\frac{h}{\beta}\left(0.43989-1.31967 \sin \left(\frac{4 \pi}{3} \gamma-\frac{\pi}{6}\right)\right)  \tag{70}\\
a=5.52794 \frac{h}{\beta^{2}} \cos \left(\frac{4 \pi}{3} \gamma-\frac{\pi}{6}\right) \tag{71}
\end{gather*}
$$

For $7 / 8 \leq \gamma \leq 1$

$$
\begin{gather*}
s=h(0.56010+0.43989 \gamma-0.035014 \sin 2 \pi(2 \gamma-1))  \tag{72}\\
v=0.43989 \frac{h}{\beta}(1-\cos 2 \pi(2 \gamma-1))  \tag{73}\\
a=5.52794 \frac{h}{\beta^{2}} \sin 2 \pi(2 \gamma-1) \tag{74}
\end{gather*}
$$

Modified Trapezoidal: This curve has been evolved by replacing the cubic equation by the cycloidal equation in the first, third, fourth and the sixth segments of the curve. Applying boundary conditions and denoting
the ratio $\theta / \beta$ by $\gamma$ the motion equations of the modified trapezoidal curve can be described as in equations 75 to 92.

For $0 \leq \gamma \leq 1 / 8$

$$
\begin{equation*}
s=0.097246 h\left(4 \gamma-\frac{1}{\pi} \sin (4 \pi \gamma)\right) \tag{75}
\end{equation*}
$$

$$
\begin{equation*}
v=0.388985 \frac{h}{\beta}(1-\cos (4 \pi \gamma)) \tag{76}
\end{equation*}
$$

$$
\begin{equation*}
a=4.888124 \frac{h}{\beta^{2}} \sin (4 \pi \gamma) \tag{77}
\end{equation*}
$$

For $1 / 8 \leq \gamma \leq 3 / 8$

$$
\begin{gather*}
s=h\left(2.4440618 \gamma^{2}-0.22204 \gamma+0.0072341\right)  \tag{78}\\
v=\frac{h}{\beta}(4.888124 \gamma-0.22204)  \tag{79}\\
a=4.888124 \frac{h}{\beta^{2}} \tag{80}
\end{gather*}
$$

For $3 / 8 \leq \gamma \leq 1 / 2$

$$
\begin{equation*}
s=h(1.6110154 \gamma-0.0309544 \sin (4 \pi \gamma-\pi)-0.3055077) \tag{81}
\end{equation*}
$$

$$
\begin{gather*}
v=\frac{h}{\beta}(1.6110154-0.388985 \cos (4 \pi \gamma-\pi))  \tag{82}\\
\left.a=4.888124 \frac{h}{\beta^{2}} \sin (4 \pi \gamma-\pi)\right) \tag{83}
\end{gather*}
$$

For $1 / 2 \leq \gamma \leq 5 / 8$

$$
\begin{equation*}
s=h(1.6110154 \gamma+0.0309544 \sin (4 \pi \gamma-2 \pi)-0.3055077) \tag{84}
\end{equation*}
$$

$$
\begin{equation*}
v=\frac{h}{\beta}(1.6110154+0.388985 \cos (4 \pi \gamma-2 \pi) \tag{85}
\end{equation*}
$$

$$
\begin{equation*}
a=-4.888124 \frac{h}{\beta^{2}} \sin (4 \pi \gamma-2 \pi) \tag{86}
\end{equation*}
$$

For $5 / 8 \leq \gamma \leq 7 / 8$

$$
\begin{equation*}
s=h\left(4.6660917 \gamma-2.44406184 \gamma^{2}-1.2292648\right) \tag{87}
\end{equation*}
$$

$$
\begin{equation*}
v=\frac{h}{\beta}(4.6660917-4.888124 \gamma) \tag{88}
\end{equation*}
$$

$$
\begin{equation*}
a=-4.888124 \frac{h}{\beta^{2}} \tag{89}
\end{equation*}
$$

$$
\begin{equation*}
s=h(0.6110154+0.388985 \gamma+0.0309544 \sin (4 \pi \gamma-3 \pi)) \tag{90}
\end{equation*}
$$

$$
\begin{gather*}
v=\frac{h}{\beta}(0.388985+0.3888985 \cos (4 \pi \gamma-3 \pi))  \tag{91}\\
a=-4.888124 \frac{h}{\beta^{2}} \sin (4 \pi \gamma-3 \pi) \tag{92}
\end{gather*}
$$

### 4.3 CAM PROFILE DETERMINATION

Cam profiles can be determined graphically or analytically. Certain cam motion curves are easily laid out by graphical procedures. Due to inherent limitations in accuracy, graphical methods are used only for relatively low-speed cams. High-speed cams require that the coordinates be calculated by a computer.

Graphical methods employ an artificial device called an 'inversion'. This represents a mental concept which is very helpful in performing the graphical work. It requires the drawing of one position of the cam and many positions $\circ$ fht $e$ follower. Instead of rotating the cam, it is assumed that the follower rotates around the 'fixed' cam. As part of the inversion process, the direction of rotation is very important. In order to preserve the correct sequence of events, the artificial rotation of the follower must be the reverse of the cam's prescribed rotation.

Because of the voluminous calculations required in conjunction with its complicated geometry, the analytical methods of determining cam profiles coordinates were subordinated to graphical techniques. This thesis uses the 'Theory of Envelopes' (3) to determine certain cam profile coordinates equations for seven major types of cam followers. We shall consider them in sequence.

In the equations shown in the following pages, $r_{b}$ is the base circle radius, $r_{f}$ the roller radius, $r_{a}$ the distance between swinger pivot point and cam center, $r_{r}$ the length of follower arm, $s$ the follower rise for angular displacement $\theta, \phi_{0}$ the intial angular postion of swinging follower, $\phi$ the change in angular position of swinging follower, $\alpha$ the pressure angle at angular displacement $\theta$, and $e$ the eccentricity of the follower.

### 4.3.1 KNIFE EDGE FOLLOWER

Figure 24 shows the cam system with the knife edge follower. In this case the point of contact between the cam and the follower always lies on the radial line joining the cam center with the knife-edge. This fact makes it very easy to compute the coordinates of the cam profile. Using the standard notations it can be seen that the rectangular coordinates of the profile are:

$$
\begin{equation*}
x=r \cos \theta \tag{93}
\end{equation*}
$$



Figure 24. Cam with Knife-Edge follower

$$
\begin{equation*}
\mathrm{y}=\mathrm{r} \sin \theta \tag{94}
\end{equation*}
$$

where,

$$
r=r_{b}+s
$$

### 4.3.2 RADIAL TRANSLATING ROLLER FOLLOWER

In this case the point of contact generally does not lie on the line joining the center of cam and the roller. The system is shown in Fig. 25. Again, the coordinates of the cam profile can be found by pure geometry. This indicates how the pressure angle $\alpha$ affects the system geometry. The $x$ and $y$ coordinates are as follows:

$$
\begin{equation*}
x=r \cos \theta-r_{f} \cos (\theta-\alpha) \tag{95}
\end{equation*}
$$

$$
\begin{equation*}
y=r \sin \theta-r_{f} \sin (\theta-\alpha) \tag{96}
\end{equation*}
$$

where,
$r=r_{b}+r_{f}+s$
$\alpha=\tan ^{-1}\left[\frac{\mathrm{ds}}{\mathrm{d} \mathrm{\theta}}\right]$


Figure 25. Cam with Roller follower

### 4.3.3 OFFSET TRANSLATING ROLLER FOLLOWER

This system is a variation of the one described above. The eccentric follower moves radially along a line that is offset from the cam center by an amount e. Equations 97 and 98 are derived from the geometry of the system shown in Fig. 26.

$$
\begin{equation*}
x=r \cos \theta-r_{f} \cos (\theta-\alpha)+e \sin \theta \tag{97}
\end{equation*}
$$

$$
\begin{equation*}
y=r \sin \theta-r_{f} \sin (\theta-\alpha)-e \cos \theta \tag{98}
\end{equation*}
$$

where,
$r=\sqrt{\left(r_{b}+r_{f}\right)^{2}-e^{2}+s}$
$\alpha=\tan ^{-1}\left[\frac{\frac{d s}{d \theta}-e}{r}\right]$

It must be noted that if $e=0$ then the above equations become the same as those for the radial follower.

### 4.3.4 SWINGING ROLLER FOLLOWER

As equations 99 and 100 show, the coordinates of the cam profile also depend on the geometry of the swing arm and the distance between the


Figure 26. Cam with Offset Roller follower

cam center and the swinger pivot point. Figure 27 shows a schematic diagram of this system.

$$
\begin{equation*}
x=r_{a} \cos \theta-r_{r} \cos \left(\theta-\phi-\phi_{0}\right)-r_{f} \sin \left(\alpha-\theta+\phi+\phi_{0}\right) \tag{99}
\end{equation*}
$$

$$
\begin{equation*}
y=r_{a} \sin \theta-r_{r} \sin \left(\theta-\phi-\phi_{O}\right)-r_{f} \cos \left(\alpha-\theta+\phi+\phi_{O}\right) \tag{100}
\end{equation*}
$$

where,
$\phi_{o}=\cos ^{-1}\left[\frac{r_{r}^{2}+r_{a}^{2}-\left(r_{b}+r_{f}\right)^{2}}{2 r_{a} r_{r}}\right]$

### 4.3.5 TRANSLATING FLAT FACED FOLLOWER

Figure 28 shows a cam with a translating flat faced follower. The general method of using geometry fails to help determine the cam profile. However combining this with the offset of the point of contact from the radial lines helps us get equations 101 and 102 for the coordinates of the cam profile.

$$
\begin{equation*}
x=\left(r_{b}+s\right) \cos \theta-\frac{d s}{d \theta} \sin \theta \tag{101}
\end{equation*}
$$



Figure 28. Cam with Flat-Faced follower

$$
\begin{equation*}
y=\left(r_{b}+s\right) \cos \theta+\frac{d s}{d \theta} \cos \theta \tag{102}
\end{equation*}
$$

### 4.3.6 SWINGING CENTRIC FLAT-FACED FOLLOWER

Figure 29 shows a swinging flat faced follower system. Using a method similar to the one used in the case of swinging roller follower, it is possible to derive equations 103 and 104 for the cam profile coordinates.

$$
\begin{equation*}
x=r_{a}(\cos \theta+A \cos \beta) \tag{103}
\end{equation*}
$$

$$
\begin{equation*}
\mathrm{y}=\mathrm{r}_{\mathrm{a}}(\sin \theta-A \sin \beta) \tag{104}
\end{equation*}
$$

where,
$A=\frac{\cos (\theta+\beta)}{\frac{d \phi}{d \theta}-1}$
$\beta=\phi_{0}+\phi-\theta$


### 4.3.7 SWINGING ECCENTRIC FLAT-FACED FOLLOWER

A swinging flat-faced follower of the eccentric type is shown in Fig. 30. The amount of eccentricity $e$ is the distance between parallel lines through the cam pivot and the follower face. The profile coordinates are:

$$
\begin{equation*}
x=r_{a}(\cos \theta+A \cos \beta)+e \sin \beta \tag{105}
\end{equation*}
$$

$$
\begin{equation*}
y=r_{a}(\sin \theta-A \sin \beta)+e \cos \beta \tag{106}
\end{equation*}
$$

where,
$A=\frac{\cos (\theta+\beta)}{\frac{d \phi}{d \theta}-1}$
$\beta=\phi_{0}+\phi-\theta$

If $e=0$ then equations 105 and 106 reduce to 103 and 104 respectively.

### 4.4 FORCE CONSIDERATIONS

The cam-follower system is subject to a number of forces. It is necessary to determine these forces and use them in the design process. The cam-follower forces are:


1. The static force due to the external load on the cam.
2. The inertia force due to follower acceleration.
3. The acceleration forces which produce vibrations.
4. Frictional forces.

Static forces: These are gradually applied forces and are characteristic of slow moving cams. The equilibrium conditions applied in static force analysis are:

$$
\begin{gathered}
F=O \text { (in the } X, Y \text {, and } Z \text { directions) } \\
M=0 \text { (about any point) }
\end{gathered}
$$

Inertia Forces: If a rotating body has an external unbalanced torque, it will have an angular acceleration which will be resisted by a reaction torque. The direction of this torque will be opposite to the direction of rotation.

$$
T=I \alpha
$$

where $I=$ Moment of Inertia about the center of rotation.

$$
\alpha=\text { angular acceleration. }
$$

Vibratory Forces: It is known that a large pulse (jerk) induces undesirable vibrations which occur at the natural frequency of the system. The forces produced by these vibrations add on to the inertia forces. Hence while designing cams for high speed applications, a dynamic magnification factor of 2 or more should be used (3).

Frictional Forces Relative movement of contacting surfaces is always opposed by frictional resistance. Friction can rarely be neglected. The best and most accurate method of including friction in the design is to use values from tests.

### 4.5 PRESSURE ANGLE CONSIDERATIONS

Pressure angle is important in cam design. Pressure angle is defined as the angle between the line of action and the center line of the follower. Pressure angles result in side thrusts on the follower and possible deflection and jamming of the follower stem. The maximum pressure angle affects the cam size, torque, loads, accelerations, wear life, and other factors. Thus the maximum pressure angle must be kept as small as possible. At present, the maximum recommended value this angle can achieve has been arbitrarily set at $30^{\circ}$. However, for light loads with accurate low-friction bearings, Rothbart(1) was successful in using pressure angles up to $47^{\circ}$.

Pressure angles can be easily measured from the graphical construction of a cam. However, analytical methods have been developed to calculate these angles for different follower types. Since solving equations for maximum pressure angles is very difficult, Mabie and Ocvirk(17) use a nomogram (Fig. 31) developed by E. C. Varnum to determine the maximum pressure angle. This chart can be used for three different types of motion. The nomogram uses $\beta_{0}$ and $L / R_{0}$ as parameters.

Pressure angles can be reduced by increasing the minimum radius of the cam so that the follower will have to travel a greater linear distance


Figure 31. Nomogram to find maximum pressure angle. (Courtesy of E. C. Varnum, Barber-Colman Company.)
on the cam for a given rise. If the workspace does not permit such an increase, then the active cam angle $\beta$ can be increased along with the rotational speed to maintain a constant lift time.

### 4.6 CAM SIZE DETERMINATION.

Care should be taken to avoid cusps or undercutting. Cusp or point formation takes place when, both, $d x / d \theta$ and $d y / d \theta$ become zero simultaneously. Based on this principle, equations have been developed to check cusp formation for different types of followers. Two simple cases are listed below:

Flat Faced Follower: If $f(\theta)$ is the displacement function for the follower and $\mathrm{f}^{\prime \prime}(\theta)$ the acceleration function, then the minimum radius of the cam (C) required, to avoid cusps or undercutting, can be easily determined by solving equation 107 for $C$.

$$
\begin{equation*}
C+f(\theta)+f^{\prime \prime}(\theta)>0 \tag{107}
\end{equation*}
$$

Radial Follower: For this type of follower, a different parameter, the radius of curvature ( $\rho$ ) of the pitch surface is considered. In order to prevent undercutting or point formation $\rho_{\text {min }}$ must be determined. Two sets of curves, that show the plot of $\rho_{\min } / R_{o}$ versus $\beta$ for different values of $L / R_{0}$, have been developed and used in Mabie \& Orvick (17). Here $\beta$ is the active cam angle, $L$ is the total lift and $R_{0}$ the minimum pitch circle
radius. Figures 32 and 33 show the graphs for Cycloidal and Harmonic motions respectively.


Figure 32. Curves for Cycloidal motion. (From M. Kloomok and R. V. Muffley,"Plate Cam Design-Radius of Curvature", Production Engineering, September 1955.)


Figure 33. Curves for Harmonic motion. (From M. Kloomok and R. V. Muffley, "Plate Cam Design-Radius of Curvature", Production Engineering, September 1955.)

## CHAPTER 5 PROGRAM DESCRIPTION

### 5.1 OVERVIEW

ANICAM has been written to allow the user to create animated sequences for disk cams. Figure 34 illustrates the input and execution flow that a user would take to design and display the cam. Upon starting ANICAM, the user will be asked to select the system on which the cam model has to be displayed. Whether the user will be able to animate the cam, depends on the choice of the display system. The two systems which have animation capability are GKS and MOVIE.BYU. Based on the design requirements the user would sequentially select the follower type (there are 7) and the basic data of the cam and the motion (displacement) characteristics . The user has a choice of 22 different displacement functions and is expected to select compatible motions. The last input asked of the user is to enter the type of output required. The user may specify any of the 3 graphics options.

ANICAM is a fairly large program and will occupy a lot of memory space. By overlaying the program, it is possible to have only a part of the program in active memory. However one shortcoming of overlaying is that it can significantly slow the program execution. To facilitate possible overlays ANICAM is broken into four major parts with the functions of program control, motion data input, motion characteristics and profile generation and display. Since ANICAM is currently running on a system with virtual memory, overlaying is not needed.


Figure 34. I/P and execution flow for ANICAM

### 5.2 PROGRAM CONTROL

The control portion of ANICAM is its heart. It calls the other program sections and passes the necessary data between them. This portion is kept small since it must always occupy active memory. The use of overlaying allows larger program segments to be called into memory above the root segment. Figure 35 illustrates a flow for the control section of the program. After starting the program the user is asked to key in the basic data for the cam and follower system. All this data is transferred through FORTRAN common blocks to the motion input program and continues on until the program has completed its operation. This process is repeated until the user indicates that he would like to stop.

### 5.3 MOTION DATA INPUT

The input section of the program takes the data regarding the displacement requirements and the functions governing the motion to be imparted to the follower. This continues until the user has specified these requirements for all the 360 degrees of the cam rotation. Once all data has been accepted, it is passed to the control section which in turn passes it to the generation section of the program. The flow chart for the input segment is shown in Fig. 36.


Figure 35. Flow of Control Section of ANICAM


Figure 36. Flow for Input Section of ANICAM

### 5.4 CAM PROFILE AND MOTION CHARACTERISTICS GENERATION

This part of the program calculates the displacement, velocity, acceleration, jerk, and the cam profile coordinates at 1 degree intervals of angular displacement of the cam. It accepts the data regarding follower motion and jumps to the proper equations for the evaluation of the motion characteristics. Data regarding the follower type, along with data just entered is passed through to the subroutine for profile calculations. Here too, the segment goes to the correct equations for cam profile coordinate calculations. Once the coordinates are evaluated all data is passed to the display segment of the program. Figure 37 shows the flowchart for the generation segment.

### 5.5 DISPLAY

This portion takes all data and generated information and executes appropriate commands for the required display system. Each display system works independently and displays the designed model using different methods. By far, this is the most important part of the program. Displaying the designed cam helps the user to decide whether the design is feasible.

If the user has chosen GKS as the display system then he is asked to enter the type of TEKTRONIX terminal on which the model is to be displayed. GKS then proceeds to display the cam profile and the motion characteristics that would be imparted to the follower if that cam model is used. Various GKS subroutines are called in order to produce a color


Figure 37. Flow for Geometry Generation Section of ANICAM
display of the cam model. A detailed list of all the subroutines called and the parameters used is given in Appendix A. GKS has the capability of performing animation. In the event that the user wishes to view the animation, GKS uses the transformation subroutine to transform the displayed segment. The illusion of animation is created by sequentially setting each transformed segment visible and invisible. The flow chart for the GKS display subroutine is shown in Fig. 38. Selected examples of the GKS display are shown in Figs. 39 and 40.

If the display system chosen is CADAM, then the user must logon to the CADAM system on the IBM 3250 or IBM 5080 before executing ANICAM. This allows the user to change the design of the cam without having to logon and run the program each time. ANICAM is linked to CADCD, a geometry interface module of the CADAM system. CADAM uses various CADCD subroutines to create and display the $3-D$ model of the cam and $2-D$ graphs of the motion characteristics. The user may perform various physical property analyses of the cam but does not have the capability to perform animation. Details of the CADCD subroutines used and the parameters used are listed in Appendix $A$. The system draws the cam profile by creating a 3-D spline between the calculated points. The flow chart for CADAM display is shown in Fig. 41. Selected examples of display obtained by this system are shown in Figs. 42 and 43.

CADAM creates a model which can be used for further design by interactively executing the CADMACGM program. This presents a new method to improve a design created by using CADAM.

If MOVIE is selected as the display system the user must leave ANICAM after creating the design. ANICAM writes the calculated data into


Figure 38. Flow. chart for display using GKS.


Figure 39. Example of GKS display.


D
i

Figure 40. Example of GKS display.


Figure 41. Flow chart for display by CADAM.


Figure 42. Example of CADAM display.


Figure 43. Example of CADAM display.

FILE.CAM, a file which has been formatted on lines of the geometry files created by the UTILITY subroutine of MOVIE.BYU. The creation of FILE.CAM enables the user to run DISPLAY without running UTILITY. The user will be asked for the name of the geometry file created using ANICAM. DISPLAY allows the model to be displayed in 3-D and be viewed at various angles. Hidden line elimination can be performed by giving a series of DISPLAY commands. If the user has chosen to animate this model while in ANICAM, he has created a command file, FILE.COM, which can be used by a modified version of DISPLAY to create and store all the frames of motion on a flexible disc. These frames can be animated by running ANIMATE (**). ANIMATE loads the data from the disc to the memory of the TEKTRONIX terminal, and sequentially displays each frame with refresh or raster graphics, giving an illusion of motion. A detailed list of all the DISPLAY commands and the format of the geometry file created are given in Appendix A. Figure 44 shows the flow chart for the MOVIE display subroutine. Selected examples of the output obtained by this display system are shown in Figs. 45 and 46.


Figure 44. Flow chart for display using MOVIE.


Figure 45. Example of MOVIE display.


Figure 46. Example of MOVIE display.

## CHAPTER 6 RECOMMENDATIONS AND IMPLICATIONS

ANICAM is intended to be an interactive program which has the capability to design and display a cam model, and the motion characteristics which will be imparted to the follower. With full color 3-D capabilities, it can be used as a powerful tool to create wire-frame or solid models by linking it with a CAD/CAM package, and a widely used graphic display package viz. CADAM and MOVIE.BYU. While CADAM allows the analysis of the physical properties of the cam designed and futher use in design or manufacturing, the ANICAM link with MOVIE and GKS can be used as one of the most powerful tool for purpose of simulation. ANICAM can be seen having many industrial applications.

Though the system has fulfilled most of the requirements of designing principles it has a tremendous potential for increased capabilities. This design is based on kinematic considerations only, and does not consider the dynamic effects on the cam. While the cam must be developed to meet motion constraints, care must be taken to see that the kinematically sound design does not fail dynamically. Dynamic effects like the loads acting on the cam, vibrations, and the stresses caused by both may also lead to the cam not being able to perform the task it was created for. In order to make the design more comprehensive and reliable, it is recommended that dynamic analysis be incorporated in the program.

The ability of the program to use $3-\mathrm{D}$ display systems brings the design of $3-D$ cams to the mind. 3-D cams are finding increasing applications in industries. The present ability of ANICAM to display 3-D
models would be highly enhanced by the capability of solid modelling. Solid modelling packages like CATIA can be easily incorporated into the program giving it the added ability to shade, giving the model a more realistic look. These abilities are expected to be available in CADAM in the near future.

The ability of the system to simulate the cam rotation would be put to better use if the follower is displayed and animated along with the cam. This would give the designer a better perspective of what to expect of the designed system and the actual design models which can be modified interactively.

One of the big advantages of the program structure is the separation of the design and display routines. By simply changing the design routine it would be possible to display other machine components like gears, linkages, pulleys, bearings, joints, chains, brakes, and clutches. It is also possible to add the design routines of the above components to ANICAM thus increasing the scope of the program.

## REFERENCES

1. Rothbart, H. H., Cams, John Wiley and Sons, 1956.
2. Jensen, P. W., Cam Design and Manufacture, The Industrial Press, 1965.
3. Chen, F. Y., Mechanisms and Design of Cam Mechanisms, Pergamon Press, 1982.
4. Johnson, R. C., "Force reduction by motion design in spring loaded cam mechanism", Journal of Mechanisms, Transmissions, and Automation in Design, Vol. 106, pp 278, 1984.
5. Donkin, W. T. and Clark, H. H., "The Electric Telemeter and Valve Spring Surge", Transactions of SAE, Vol. 24, pp 687-700, 1979.
6. Pisano, A. P. and Freudenstein, F., "An experimental and analytical investigation of the dynamic response of the high speed cam follower system", Journal of Mechanisms, Transmissions, and Automation in Design, Vol. 105, pp 692-704, 1984.
7. Berzak N., "Optimization of cam follower system in polydyne cam design", Doctoral thesis, Columbia University, 1979.
8. Sermon, C. F. and Liniecki, A., "Search for optimum solution of a single disk cam mechanism with an oscillating roller follower", ASME Pap. 72-MECH-61, 1972.
9. Sandler, B. Z., "Suboptimal dynamic synthesis of linearized mechanical system", ASME Pap. 74-DET-27, 1974.
10. Chen, F. Y. and Shah, A. M., "Optimal design of geometric parameters of a cam mechanism using sequential random vector technique", ASME Pap. 72-Mech-73, 1972.
11. Chew, M., Freudenstein, F., and Longman, R. W., "Application of optimal control theory to the synthesis of high speed cam follower systems", Journal of Mechanisms, Transmissions, and Automation in Design Vol. 105, pp 576-591, 1984.
12. Lafuente, J. M., "Interactive graphics in data processing cam design on graphics console", IBM Syst. J:3-4, pp 325-343, 1968.
13. De Fraine, J., "Integration of computer aided design and computer aided manufacturing for cam driving mechanisms", Proceedings of 5 th world congress on theory of machines and mechanisms, pp 122-125, 1969.
14. Tesar, D., and Mathew, G. K., The dynamic synthesis, analysis and design of modeled cam systems, Lexington Books, Lexington, Massechusetts, 1976.
15. Vallard, H., "Computer programs for analysis of valve train dynamics and cam profile evaluation", The Institute of Internal Combustion Engines Bulletin No. IF-H5, Trondheim, Norway, 1969.
16. Keil, M. J., "A Method for Automatically Generating and Animating 3-D Models of Planar Linkages", Master's thesis, Florida Atlantic University, 1984.
17. Mabie, H. H., and Ocvirk F. W., Mechanisms and Dynamics of Machinery, John Wiley and Sons, 1978.
18. Ballaney, P. L., Theory of Machines, Khanna Publishers, New Delhi, 1980.
19. Tausworthe, R. C., Standardized Development of Computer Software
(Part 1), Prentice-Hall Inc., 1977 .
20. Foley, J. D., Vam Dam, A., Fundamentals of Interactive Computer Graphics, Addison-Wesley Publishing Company, 1984.
21. Sproull, R. F., Sutherland, W. R., Ullner, M. K., Device-Independent Graphics, McGraw-Hill Book Company, 1985.
22. Hopgood, F. R. A., Duce, D. A., Gallop, J. R., Sutcliffe, D. C., Introduction to the Graphical Kernel System, Academic Press, 1983.
23. Myklebust, A., "ANIMATE", Florida Atlantic University, 1983.
24. Brigham Young University, MOVIE.BYU.-- General Purpose Computer Graphics System, Depatment of Computer Science, 1984.
25. CADAM Inc., Computer-Graphics Augmented Design and Manufacturing System - User Training Manual SH20-2035-3, IBM, 1982.
26. CADAM Inc., CADAM System - Geometry Interface Module Installation and Programmers Guide SH20-2099-6, IBM, 1983. Guide SH20-6227-0, IBM, 1985.

## APPENDIX A.

## Subroutines used for GKS.

1. Activate workstation

CALL GACWK (WKID)
WKID = workstation identifier.
2. Close GKS

CALL GCLKS
3. Close segment

CALL GCLSG
4. Close workstation

CALL GCLWK (WKID)
WKID $=$ workstation identifier.
5. Create segment

CALL GCRSG (SGNA)
SGNA $=$ segment name
6. De-activate workstation

CALL GDAWK (WKID)
WKID $=$ workstation identifier.
7. Evaluate transformation matrix

CALL GEVTM (XO, YO, DX, DY, PHI, FX,FY,SW,MOUT)
$\mathrm{XO}=\mathrm{x}$ coordinate of the fixed point
YO $=\mathrm{y}$ coordinate of the fixed point
DX = shift vector x coordinate
$D Y=$ shift vector $y$ coordinate
PHI $=$ Rotation angle in radians
$\mathrm{FX}=\mathrm{x}$ coordinate scale factor
$F X=x$ coordinate scale factor
SW = Coordinate system
MOUT $=$ Resulting transformation matrix
8. Draw fill area

CALL GFA(N,PX,PY)
$\mathrm{N}=$ Number of points
$\operatorname{PX}(N)=x$ coordinates of points in WC
$P Y(N)=y$ coordinates of points in WC
9. Open GKS

CALL GOPKS (ERRFIL)
ERRFIL = Error message file.
10. Open workstation

CALL GOPWK (WKID,CONID,WTYPE)
WKID = workstation identifier.
CONID = Connection identifier
WTYPE = Workstation type
11. Draw polyline

CALL GPL(N,PX,PY)
$\mathrm{N}=$ Number of points
$\mathrm{PX}(\mathrm{N})=\mathrm{x}$ coordinates of points in WC
$\mathrm{PY}(\mathrm{N})=\mathrm{y}$ coordinates of points in WC
12. Draw polymarker

CALL GPM(N,PX,PY)
$\mathrm{N}=$ Number of points
$\operatorname{PX}(\mathrm{N})=\mathrm{x}$ coordinates of points in WC
$P Y(N)=y$ coordinates of points in WC
13. Set character height.

CALL GSCHH (CHH)
$\mathrm{CHH}=$ Character height in WC
14. Set character expansion factor.

CALL GSCHXP (CHXP)
CHXP = Character expansion factor
15. Set normalization transformation.

CALL GSELNT(TNR)
TNR $=$ Transformation identifier
16. Set fill area color index.

CALL GSFACI (COLI)
COLI $=$ Integer number corresponding to one color
17. Set fill area interior style.

CALL GSFAIS(INST)
INST = Integer number corresponding one style
18. Set line type.

CALL GSLN((LTYP)
LTYP $=$ Integer number corresponding to one line type
19. Set marker type.

CALL GSMK (MTYPE)
MTYP $=$ Integer number corresponding to one marker type
20. Set polyline color index.

CALL GSPLCI (COLI)

COLI $=$ Integer number corresponding to one color
21. Set polymarker color index.

CALL GSPMCI (COLI)

COLI $=$ Integer number corresponding to one color
22. Set segment transformation.

CALL GSSGT(SGNA,M)

SGNA $=$ Segment name
$M=$ Transformation matrix
23. Set text alignment.

CALL GSTXAL(TXALH,TXALV)
TXALH $=$ Horizontal alignment of text

TXALV $=$ Vertical alignment of text
24. Set text color index.

CALL GSTXCI (COLI)

COLI $=$ Integer number corresponding to one color
25. Set text font and precision.

CALL GSTXFP (FONT, PREC)

FONT $=$ Integer corresponding to one font

PREC $=$ Integer corresponding to one precision value
26. Set segment visibility.

CALL GSVIS (SGNA,I)

SGNA $=$ Segment name
I = 0 for invisible, 1 for visible
27. Set viewport.

CALL GSVP (TNR,XMIN,XMAX,YMIN,YMAX)

TNR = Transformation identifier

XMIN = Minimum value of $x$ in NDC coordinates $X M A X=$ Maximum value of $x$ in NDC coordinates YMIN $=$ Minimum value of $y$ in NDC coordinates YMIN = Maximum value of $y$ in NDC coordinates
28. Set workstation viewport.

CALL GSWKVP (WKID, XMIN, XMAX, YMIN, YMAX)
WKID $=$ Workstation identifier

XMIN $=$ Minimum value of $x$ in $D C$ coordinates

XMAX = Maximum value of $x$ in DC coordinates

YMIN $=$ Minimum value of $y$ in $D C$ coordinates
YMIN $=$ Maximum value of $y$ in DC coordinates
29. Set workstation window.

CALL GSWKWN(WKID,XMIN,XMAX,YMIN,YMAX)
WKID $=$ Workstation identifier

XMIN $=$ Minimum value of $x$ in NDC coordinates
$X M A X=$ Maximum value of $x$ in NDC coordinates
YMIN $=$ Minimum value of $y$ in NDC coordinates
YMIN $=$ Maximum value of $y$ in NDC coordinates
30. Set window.

- CALL GSWN(TNR,XMIN, XMAX,YMIN, YMAX)

TNR $=$ Transformation number

XMIN $=$ Minimum value of $x$ in WC coordinates
$X M A X=$ Maximum value of $x$ in WC coordinates
YMIN = Minimum value of $y$ in $W C$ coordinates
YMIN $=$ Maximum value of $y$ in $W C$ coordinates
31. Draw text.

CALL GTX ( $($ PX, PY,CHARS $)$
$P X=x$ coordinate of text postion in WC
PY $=\mathrm{y}$ coordinate of text postion in WC
CHARS $=$ String of characters
32. Update workstation.

CALL GUWK (WKID,REGFL)
WKID = Workstation identifier
REGFL $=$ Extent of update functions

Subroutines used for CADAM.

1. Begin A Trap for View Geometry

CALL BEGVU(IOPT,IDVU,XYZ,XVCTR,YVCTR,ZVCTR,*ERROR)
IOPT = Option flag which must be set to one. This flag eventually determines the handling of input coordinates.

IDVU $=4$ characters, of which the last two are used 'xxID'
$X Y Z=A n$ array $X Y Z(3)$ representing the $X, Y$, and $Z$ location of the view.

XVCTR $=$ An array XVCTR(3) representing the direction cosines of the new $X$-axis.

YVCTR $=$ An array YVCTR(3) representing the direction cosines of the new $Y$-axis.

ZVCTR $=$ An array $Z V C T R(3)$ representing the direction cosines of the new 2 -axis.
*ERROR $=$ ERROR. A view has already been started, or no more space for views.
2. File the Drawing

CALL CADFIL(IOPT,NOGOOD,IDUMMY)
$I O P T=0$ File a drawing.
1 Retrieve a drawing.
2 Overfile an drawing.
NOGOOD $=$ An integer to be tested after filing. If not equal to one, the file call was no good.

IDUMMY = Dummy varailble for later use.
3. Initialize a model.

CALL CADST(DRAWID,EMPNO,GROUP)
DRAWID $=5$ word array containg drawid.
EMPNO $=2$ word array containing the employee ID.
GROUP $=$ One word array containing the group ID.
4. Draw a 3-D line.

CALL LINE3D (XYZ1,XYZ2,*ERROR)
XYZ1 = An array(3) containing the X, Y, Z absolute coordinates of one of the end points of the line.
$\mathrm{XYZ2}=$ An array (3) containing the $\mathrm{X}, \mathrm{Y}, \mathrm{Z}$ absolute coordinates of the other end point of the line.
*ERROR $=$ Error return indicating that the length of the line is less than the tolerance limit of 0.001 .
5. Draw a 3-D Spline.

CALL SPLINE (N,M,K,D,S,T1,T2,T3,*ERROR)
$\mathrm{N}=$ Number of input points
$M=$ Number of dimensions in space. (1,2 or 3 )
$D(M, N)=A$ floating point array containing spline points coordinate data.
$V(M, N)=A$ floating point array of spline tangent vector components. $S(N)=A$ floating point array of chord lengths for bays defined by the input data.

T1 = Temporary floating point working area, must be of length N .
$\mathrm{T} 2=$ Temporary floating point working area, must be of length N .
T3 = Temporary floating point working area, must be of length N . $\therefore E R R O R=$ Denotes label in the program which subroutine returns in case of an error.

## Writing the Geometry file for MOVIE.BYU.

FILE.CAM generated by ANICAM is written in the same format as the geometry file generated by using the UTILITY program of MOVIE.BYU. The format used in writing the geometry file is as follows:

```
If the variables used are defined as
NP = Number of parts
\(\mathrm{NJ}=\) Number of nodes or joints
NPT = Number of elements or polygons
NCON = Number of entries in the connectivity array
NPL \(=\) The parts list
\(X=\) The coordinates of the nodes
```

$I P=$ The connectivity of the elements or polygons
then the geometry file is written using the following fortran statements:

WRITE $(10,100)$ NP,NJ,NPT,NCON
$\operatorname{WRITE}(10,100)((\operatorname{NPL}(\mathrm{I}, \mathrm{IJ}), \mathrm{I}=1,2), \mathrm{IJ}=1, \mathrm{NP})$
$\operatorname{WRITE}(10,200)((X(I, I J), I=1,3), I J=1, N J)$
WRITE $(10,100)(I P(I), I=1, N C O N)$
100 FORMAT (16I5)
200 FORMAT (6E12.5)

More detailed information can be got by referring to any of the manuals available on GKS, CADAM, and MOVIE.BYU.

## APPENDIX B.

This is the exec for ANICAM. This exec asks the user to choose the display system on which the cam model is to be displayed. It allows the user to QUIT from the program any time before the execution begins.

## \&TRACE OFF

FILEDEF 7 DISK CAM OUT A
FILEDEF 8 DISK FILE CAM A
FILEDEF 9 DISK FILE COM A
-5
CLEAR
\&LOOP 112
\&TYPE
\&BEGTYPE 8
SELECT DISPLAY SYSTEM

1. GKS
2. CADAM
3. MOVIE.BYU
4. NO DISPLAY

Type QUIT to exit from program.
\&READ VARS \&MONI
CLEAR
\&IF .\&MONI EQ . 1 \&GOTO -10
\&IF .\&MONI EQ . 2 \&GOTO -20
\&IF .\&MONI EQ . 3 \&GOTO -30
\&IF .\&MONI EQ . 4 \&GOTO -40
\&IF . \&MONI EQ .QUIT \&GOTO -50
\&GOTO -5
-10
\&LOOP 110
\&TYPE
\&BEGTYPE 7
Hit RETURN to load for GKS.
Execution will begin 30 seconds after hitting RETURN key.
If selection is to be changed, type INVALID to return to menu.

Type QUIT to exit from the program.
\&READ VARS \&DUMB
\&IF .\&DUMB EQ . INVALID \&GOTO -5
\&IF .\&DUMB EQ .QUIT \&GOTO -50
LOAD ANICAM1 (START

```
&GOTO -50
-20
&LOOP 1 10
&TYPE
&BEGTYPE }
```

Hit RETURN to load for CADAM.
Execution will begin 45 seconds
after hitting RETURN key.
If selection is to be changed,
type INVALID to return to menu.

Type QUIT to exit from the program.
\&READ VARS \&DUMB
\&IF . \&DUMB EQ . INVALID \&GOTO -5
\&IF . \&DUMB EQ .QUIT \&GOTO -50
EXEC CADCDLNK ANICAM2
EXEC CADCD MECH ANICAM2
\&GOTO -50
-30
\&LOOP 110
\&TYPE
\&BEGTYPE 7
Hit RETURN to load for MOVIE.BYU. Execution will begin 5 seconds after hitting RETURN key. If selection is to be changed, type INVALID to return to menu.

Type QUIT to exit from the program.
\&READ VARS \&DUMB
\&IF .\&DUMB EQ .INVALID \&GOTO -5
\&IF .\&DUMB EQ .QUIT \&GOTO -50
LOAD ANICAM3 (START
\&GOTO -50
-40
\&LOOP 110
\&TYPE
\&BEGTYPE 7
Hit RETURN to load for SCREEN.
Execution will begin 5 seconds
after hitting RETURN key.
If selection is to be changed, type INVALID to return to menu.

Type QUIT to exit from the program.
\&READ VARS \&DUMB
\&IF .\&DUMB EQ . INVALID \&GOTO -5
\&IF . \&DUMB EQ .QUIT \&GOTO -50
LOAD ANICAM4 (START
-50
CLEAR
\&EXIT

## APPENDIX C.



This is the alphabetical list of the main variables that have been used in the program, ANICAM.

1. A: An array of 361 real numbers representing the acceleration of the follower at one degree angular displacement of the cam.
2. ALPHA: An array of 361 real numbers representing the pressure angle at one degree angular displacement of the cam.
3. B: Parameter with value $1125 . / 1192$.
4. BASE: This is a real number representing the base radius of the cam.
5. BETA: This is a real number representing the active cam angle for each cam segment. (radians)
6. C: Integer number representing the UPTO value of the $M-1$ th segment.
7. DESI: This is an integer number representing the design technique to be adopted.
8. E: An array of $N$ integer numbers representing the type of curve selected.
9. ECC: This is a real number representing the eccentricity of the follower with the cam center.
10. FROM: An array of $N$ integer numbers representing the start of each cam segment. (degrees)
11. GAMMA: This is a real number representing the ratio THETA/BETA.
12. $H:$ An array of $N$ real numbers representing the amount of lift, drop, dwell or swing of the follower.
13. I: Counter
14. J: An array of 361 real numbers representing the jerk on the follower at one degree angular displacement of the cam.
15. L: Counter used for GKS.
16. M: Counter used for computing the motion characteristics.
17. MONI: Integer number used for selecting the type of display system.
18. N : Integer number used to represent number of segments in the cam.
19. OUT: This is an integer number representing the parameters to be displayed on the monitor.
20. PI: Parameter with value 3.1415926
21. R: Integer number representing selection of nature of design changes to be made.
22. RA: This is a real number representing the distance between the cam center and the swinging follower.

23．RF：This is a real number representing the length of the follower arm．

24．ROLL：This is a real number representing the radius of the roller follower．

25．RPM：This is an integer number representing the rotational speed of the cam in rotations per minute．

26．S：An array of 361 real numbers representing the displacement of the follower at one degree angular displacement of the cam．

27．THETA：This is a real number representing the angular dispalcement of the cam．（radians）

28．TYPE：This is an integer number representing the type of follower selected．

29．UPTO：An array of $N$ integer numbers representing the start of each cam segment．（degrees）

30．V：An array of 361 real numbers representing the velocity of the follower at one degree angular displacement of the cam．

31．WIDTH：This is a real number representing the width of the cam．
32．$X:$ An array of 361 real numbers representing the $x$－coordinate of the cam profile at one degree interval．

33．Y：An array of 361 real numbers representing the $y$－coordinate of the cam profile at one degree interval．

34．Z：An array of 361 real numbers representing the z－coordinate of the cam profile at one degree interval．

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| ＋ |  | 大市 |





产：THE MAIN PORTION OF THE PROGRAM OPERATES AS THE NERVE CENTER＊大 OF THE SYSTEM．IT CALLS VARIOUS SUBROUTINES TO DESIGN THE＊
雨 CAM．IT ALLOWS REPEATED CHANGES IN THE DESIGN TILL THE USER＊
夫 IS SATISFIED WITH THE DESIGN OR DECIDES TO QUIT．THIS PART＊
产 OF THE PROGRAM ALSO ALLOWS THE USER TO SAVE THE OUTPUT IN 水六
犬 FORM OF AN OUTPUT FILE CALLED CAM OUT．六六

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COMMON／CAM1／TYPE，BASE，RPM，WIDTH，DESI，ROLL，ECC，RA，RF
COMMON／COUNT／I，N，M，B，PI，C，R，L
COMMON／ANGLE／BETA，THETA，GAMMA
COMMON／MTN／E（25），H（20），FROM（20），UPTO（20）
COMMON／CRV／S（360），V（360），A（360），J（360）
COMMON／COOD／ALPHA（360），X（361），Y（361），Z（361）
COMMON／DISP／OUT，MONI
INTEGER I，N，M，C，R
INTEGER TYPE，OUT，DESI，MONI
INTEGER E，FROM，UPTO
REAL BETA，THETA，GAMMA，B，PI，BASE，RPM，WIDTH
REAL ROLL，ECC，RA，RF
REAL S，V，A，J
REAL ALPHA，X，Y，Z，H

CHARACTER CHAR
PI $=3.1415926$
$B=1125 . / 1192$ ．
$\mathrm{L}=0$
10 FORMAT（A1）
100 CALL FOLLOW
CALL BASIC
CALL OUTPUT
CALL DESIGN
CALL MOTION
CALL DISPLY
CALL CURVES
CALL PROFIL
300 WRITE $(6, *)^{\prime} \quad$ Do you want to change the design？（Y／N）＇
$\operatorname{READ}(5,10)$ CHAR

IF (CHAR .EQ. 'Y') THEN
CALL CHANGE
GOTO 200
ENDIF
$\operatorname{WRITE}(6, *){ }^{\prime}$
$\operatorname{READ}(5,10)$
$\operatorname{CHAR}$
IF (CHAR .EQ. 'Y') CALL DOCUM
$\operatorname{WRITE}(6, *) \quad$ ' Do you want to create a new Design? (Y/N)'
$\operatorname{READ}(5,10)$ CHAR $^{\prime}$
IF (CHAR .EQ. 'Y') GOTO 100
$\operatorname{WRITE}(6, *)$ ' Are you ready to quit? (Y/N)'
$\operatorname{READ}(5,10)$ CHAR
IF (CHAR .NE. 'Y') GOTO 300
STOP
END


SUBROUTINE FOLLOW
COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF
COMMON/COUNT/ I, N, M, B, PI, C, R, L
COMMON/ANGLE/ BETA, THETA, GAMMA
COMMON/MTN/ E(25), H(20), FROM(20), UPTO(20)
COMMON/CRV/ S(360), V(360), A(360), J(360)
COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)

COMMON/DISP/ OUT, MONI
INTEGER I, N, M, C, R
INTEGER TYPE, OUT, DESI, MONI
INTEGER E, FROM, UPTO
REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
REAL ROLL, ECC, RA, RF
REAL S, V, A, J
REAL ALPHA, X, Y, Z, H
100 WRITE $(6, *)$
$\operatorname{WRITE}(6, *)$
$\operatorname{WRITE}(6, *)$
$\operatorname{WRITE}(6, *)$ 1: Knife Edge'
WRITE (6,*) ' 2: Radial Translating Roller'
WRITE $(6, *)$ ' 3: Offset Translating Roller'
$\operatorname{WRITE}(6, *)$
' 4: Swinging Roller'
WRITE (6,*)
5: Translating Flat Faced'
WRITE (6,*)
6: Swinging Centric Flat-Faced'
WRITE $(6, *)$
7: Swinging Eccentric Flat-Faced'
$\operatorname{READ}(5, *)$ TYPE
Please Enter Follower Type'
'

IF (TYPE .LT. 1 .OR. TYPE .GT. 7) THEN
WRITE $(6, *)$ ERROR!! INVALID SELECTION!' GOTO 100
ENDIF
IF (TYPE .GT. 1 .AND. TYPE .LT. 5) THEN
$\operatorname{WRITE}(6, *)$ ' Please enter roller radius' $\operatorname{READ}(5, *)$ ROLL

IF (ROLL .LT. 0) THEN
WRITE $(6, *)^{\prime} \quad$ ERROR!! RADIUS MUST BE POSITIVE'
ENDIF
ENDIF
IF (TYPE .EQ. 3 .OR. TYPE .EQ. 7) THEN
WRITE $(6, *)^{\prime} \quad$ Please enter eccentricity'
$\operatorname{READ}(5, *) \operatorname{ECC}$
ENDIF
IF (TYPE .EQ. 4 .OR. TYPE .EQ. 6 .OR. TYPE .EQ. 7) THEN
300

Enter center distance between'
Cam Center and Follower Arm.'
$\operatorname{READ}(5, *)$ RA

IF (RA .LT. 0) THEN
$\operatorname{WRITE}(6, *)$
ERROR!! DISTANCE MUST BE POSITIVE'
GOTO 300
ENDIF

## ENDIF

IF (TYPE .EQ. 4 .OR. TYPE .EQ. 6) THEN
$400 \operatorname{WRITE}(6, *)$
$\left.\operatorname{READ}(5,)^{\prime}\right) \operatorname{RF} \quad \quad$ Enter length of the Follower Arm'

IF (RF .LT. 0) THEN<br>WRITE (6,*) ' ERROR!! LENGTH MUST BE POSITIVE'<br>GOTO 400<br>ENDIF

ENDIF
RETURN
END


## SUBROUTINE BASIC

COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF COMMON/COUNT/ I, N, M, B, PI, C, R, L COMMON/ANGLE/ BETA, THETA, GAMMA
COMMON/MTN/ E(25), H(20), FROM(20), UPTO (20)
COMMON/CRV/ S(360), V(360), A(360), J(360)
COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
COMMON/DISP/ OUT, MONI
INTEGER I, N, M, C, R
INTEGER TYPE, OUT, DESI, MONI
INTEGER E, FROM, UPTO
REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH

```
    REAL ROLL, ECC, RA, RF
    REAL S, V, A, J
    REAL ALPHA, X, Y, Z, H
100 WRITE (6,*)'
    READ(5,*) BASE
    IF(BASE .LT. 0) THEN
        WRITE (6,*)
        GOTO 100
    ENDIF
200 WRITE (6,*)'
    READ(5,*) RPM
    IF(RPM .LT. 0) THEN
        WRITE (6,*)
        GOTO 200
    ENDIF
300 WRITE(6,*)'
    READ(5,*) WIDTH
    IF(WIDTH .LT. 0) THEN
            WRITE (6,*)' ERROR!! WIDTH MUST BE POSITIVE'
            GOTO 300
    ENDIF
    RETURN
    END
```



## SUBROUTINE OUTPUT

```
            COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF
            COMMON/COUNT/ I, N, M, B, PI, C, R, L
            COMMON/ANGLE/ BETA, THETA, GAMMA
            COMMON/MTN/ E(25), H(20), FROM(20), UPTO(20)
            COMMON/CRV/ S(360), V(360), A(360), J(360)
            COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
            COMMON/DISP/ OUT, MONI
            INTEGER I, N, M, C, R
            INTEGER TYPE, OUT, DESI, MONI
            INTEGER E, FROM, UPTO
            REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
            REAL ROLL, ECC, RA, RF
            REAL S, V, A, J
            REAL ALPHA, X, Y, Z, H
100 WRITE (6,*) ' Please select type of Output'
WRITE (6,*) ' '
WRITE (6,*)'
WRITE (6,*) ' 1: Profile only'
WRITE(6,*) ' 2: Curves only'
WRITE(6,*) ' 3: Profile and Curves'
READ (5,*) OUT
IF (OUT .LT. 1 .OR. OUT .GT. 3) THEN
                                    WRITE(6,*) ' ERROR!! INVALID SELECTION!'
                                    GOTO 100
ENDIF
RETURN
END
```

| 6\％ |  |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 六穴－＊＊＊＊ |  |  |  |  |  |  |  |  |  |  |  |
| \％ | SUBROUTINE DESIGN＊＊＊＊ |  |  |  |  |  |  |  |  |  |  |
| 交穴 寺安 |  |  |  |  |  |  |  |  |  |  |  |
| 言 <br>  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
| \％－¢冖 |  |  |  |  |  |  |  |  |  |  |  |
| 兴 THIS SUBROUTINE ASKS THE USER TO SELECT THE DESIGN PROCESS 安 |  |  |  |  |  |  |  |  |  |  |  |
| \％TO BE ADOPTED IN ORDER TO DESIGN THE CAM．ONLY THE FIRST |  |  |  |  |  |  |  |  |  |  |  |
| 大\％ | OPTION IS OPERATING AT THIS MOMENT．THIS SUBROUTINE WILI 交d |  |  |  |  |  |  |  |  |  |  |
| \％ | ALIOW THE OPTIMIZATION PROCESS TO BE INCORPORATED IN THE ．． |  |  |  |  |  |  |  |  |  |  |
| \％ | PROGRAM WITHOUT MANY COMPLICATIONS． |  |  |  |  |  |  |  |  |  |  |
| 市\％ |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |

```
SUBROUTINE DESIGN
COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF
COMMON/COUNT/ I, N, M, B, PI, C, R, L
COMMON/ANGLE/ BETA, THETA, GAMMA
COMMON/MTN/ E(25), H(20), FROM(20), UPTO(20)
COMMON/CRV/ S(360), V(360), A(360), J(360)
COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
COMMON/DISP/ OUT, MONI
INTEGER I, N, M, C, R
INTEGER TYPE, OUT, DESI, MONI
INTEGER E, FROM, UPTO
REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
REAL ROLL, ECC, RA, RF
REAL S, V, A, J
REAL ALPHA, X, Y, Z, H
100 WRITE (6,*); , Please select Design Process'
IF (DESI .LT. 1 .OR. DESI .GT. 2) THEN
    WRITE(6,*) ' ERROR!! INVALID SELECTION!'
    GOTO 100
ENDIF
RETURN
END
```



SUBROUTINE MOTION
COMMON／CAM1／TYPE，BASE，RPM，WIDTH，DESI，ROLL，ECC，RA，RF
COMMON／COUNT／I，N，M，B，PI，C，R，L
COMMON／ANGLE／BETA，THETA，GAMMA
COMMON／MTN／E（25），H（20），FROM（20），UPTO（20）
COMMON／CRV／S（360），V（360），A（360），J（360）
COMMON／COOD／ALPHA（360），X（361），Y（361），Z（361）
COMMON／DISP／OUT，MONI
INTEGER I，N，M，C，R
INTEGER TYPE，OUT，DESI，MONI
INTEGER E，FROM，UPTO
REAL BETA，THETA，GAMMA，B，PI，BASE，RPM，WIDTH
REAL ROLL，ECC，RA，RF
REAL S，V，A，J
REAL ALPHA，X，Y，Z，H

```
    I = 0
100 I = I + 1
    WRITE(6,*)' Please enter Lift/Drop/Dwell/Swing'
    READ(5,*) H(I)
    IF (TYPE .EQ. 4 .OR. TYPE .EQ. 6 .OR. TYPE .EQ. 7) THEN
        H(I) = H(I)*PI/180.
    ENDIF
200 WRITE (6,*)
Please Enter Duration'
WRITE (6,*)'
    READ(5,*) FROM(I),UPTO(I)
    IF (UPTO(I) .GT. 360) THEN
        WRITE(6,*) ' ERROR!! DURATION MUST BE LESS THAN 360'
        GOTO 200
    ENDIF
    IF (H(I) .EQ. 0) THEN
        E(I) = 25
        GOTO 400
    ENDIF
300 WRITE (6,*)'
    WRITE(6,*)' '
    WRITE (6,*) ' 1: H-1 SHM'
    WRITE (6,*)' 2: H-2 SHM'
    WRITE (6,*)
    3:
    H-3 SHM'
```

| WRITE (6,*) | 4: | $\begin{array}{ll} \text { H-4 SHM' } \\ \text { H-5 SHM' } \end{array}$ |  |  |
| :---: | :---: | :---: | :---: | :---: |
| WRITE ( $6, *$ ) | 5: |  |  |  |
| WRITE (6,*) | 6: | H-6 SHM ${ }^{\prime}$ |  |  |
| WRITE (6,*) | 7: | C-1 Cycloidal' |  |  |
| WRITE (6,*) | 8: | C-2 Cycloidal' |  |  |
| WRITE $(6, *)$ | $9:$ | C-3 Cycloidal' |  |  |
| WRITE (6,*) | 10: | C-4 Cycloidal' |  |  |
| $\operatorname{WRITE}(6, *)$ | 11: | C-5 Cycloidal' |  |  |
| WRITE (6,*) | 12: | C-6 Cycloidal' |  |  |
| WRITE (6,*) | $13:$ | Constant Velocity' |  |  |
| WRITE (6,*) | 14: | Constant Accl/Dccl' |  |  |
| WRITE ( $6, *$ ) | $15:$ | 2-3 Polynomial' |  |  |
| WRITE (6,*) | 16: | 3-4-5 Polynomial' |  |  |
| WRITE (6,*) | 17 : | 4-5-6-7 Polynomial' |  |  |
| WRITE (6,*) | 18: | Gutman 1-3 Harmonic (90)' |  |  |
| WRITE (6,*) |  | Freudenstein 1-3 Harmonic(84)' |  |  |
| WRITE (6,*) |  | Freudenstein 1-3-5 Harmonic(84)' |  |  |
| WRITE (6,*) | 21: | Modified Sine' |  |  |
| WRITE (6,*) | 22: | Modified Trapezoidal' |  |  |
| $\operatorname{READ}(5, *) \mathrm{E}(\mathrm{I})$ |  |  |  |  |
| $\begin{aligned} & \text { IF (E(I) .LT. } 1 \\ & \begin{array}{l} \text { WRITE }(6, *) \\ \text { GOTO } 300 \end{array}, \quad \text { OR. E(I) .GT. 22) THEN } \\ & \text { ERROR!! INVALID SELECTION!' } \end{aligned}$ |  |  |  |  |
| ENDIF |  |  |  |  |
| IF (UPTO(I) .LT. 360) GOTO 100 |  |  |  |  |
| $\mathrm{N}=\mathrm{I}$ |  |  |  |  |
| RETURN |  |  |  |  |
| END |  |  |  |  |


 ب篗

SUBROUTINE DISVEL
COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF COMMON/COUNT/ I, N, M, B, PI, C, R, L COMMON/ANGLE/ BETA, THETA, GAMMA
COMMON/MTN/ E(25), H(20), FROM(20), UPTO (20)
COMMON/CRV/ S(360), V(360), A(360), J(360)
COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
COMMON/DISP/ OUT, MONI
INTEGER I, N, M, C, R
INTEGER TYPE, OUT, DESI, MONI
INTEGER E, FROM, UPTO
real beta, theta, gamma, B, PI, BASE, RPM, WIDTH
REAL ROLL, ECC, RA, RF
REAL S, V, A, J
REAL ALPHA, X, Y, Z, H
DO $4000 \mathrm{M}=1, \mathrm{~N}$
BETA $=($ UPTO $(M)-U P T O(M-1)) * P I / 180$.
$\mathrm{C}=\mathrm{UPTO}(\mathrm{M}-1)$
THETA $=0$.
GO TO $(100,200,300,400,500,600,700,800,900,1000,1100,1200,1300$,
\# $\quad 1400,1500,1600,1700,1800,1900,2000,2100,2200), E(M)$
*
DO 11 I = FROM(M), UPTO(M)
THETA $=$ THETA $+($ PI/180.)
GAMMA $=$ THETA/BETA
$S(I)=S(C)$
$V(I)=0$.
11 CONTINUE
GOTO 4000

100 DO $101 \mathrm{I}=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+($ PI/180.)
GAMMA $=$ THETA $/$ BETA
$S(I)=S(C)+H(M) *(1-\operatorname{COS}(P I * G A M M A / 2))$
$\mathrm{V}(\mathrm{I})=((\mathrm{PI} / 2) * \mathrm{H}(\mathrm{M}) / \mathrm{BETA}) *(\mathrm{SIN}(\mathrm{PI} *$ GAMMA/2)$)$
101 CONTINUE
GOTO 4000


```
200 DO 201 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    S(I) = S(C) + H(M)*(SIN(PI*GAMMA/2))
    V(I) = ((PI/2)*H(M)/BETA )*(COS(PI*GAMMA/2))
201 CONTINUE
GOTO 4000
```


300 DO $301 \mathrm{I}=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+($ PI $/ 180$.
GAMMA $=$ THETA/BETA
$S(I)=-H(M) *(\operatorname{COS}(P I * G A M M A / 2))$
$V(I)=((P I / 2) * H(M) / B E T A) *(S I N(P I * G A M M A / 2))$
301 CONTINUE
GOTO 4000

400 DO $401 \mathrm{I}=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+(\mathrm{PI} / 180$.
GAMMA $=$ THETA $/$ BETA
$S(I)=-H(M) *(1-S I N(P I * G A M M A / 2))$
$V(\mathrm{I})=((\mathrm{PI} / 2) * \mathrm{H}(\mathrm{M}) /$ BETA $) *(\mathrm{COS}(\mathrm{PI} *$ GAMMA $/ 2))$
401 CONTINUE
GOTO 4000

500 DO $501 \mathrm{I}=\mathrm{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+($ PI/180.)
GAMMA $=$ THETA $/$ BETA
$\mathrm{S}(\mathrm{I})=\mathrm{S}(\mathrm{C})+(\mathrm{H}(\mathrm{M}) / 2) *(1-\operatorname{COS}(\mathrm{PI} * G A M M A))$
$\mathrm{V}(\mathrm{I})=((\mathrm{PI} / 2) * \mathrm{H}(\mathrm{M}) / \mathrm{BETA}) *(\mathrm{SIN}(\mathrm{PI} * \mathrm{GAMMA}))$
501 CONTINUE
GOTO 4000

600 DO 601 I $=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+($ PI/180.)
GAMMA $=$ THETA/BETA
$S(I)=-(H(M) / 2) *(1+\operatorname{COS}(P I * G A M M A))$
$\mathrm{V}(\mathrm{I})=((\mathrm{PI} / 2) * \mathrm{H}(\mathrm{M}) /$ BETA $) *(\mathrm{SIN}(\mathrm{PI} * \mathrm{GAMMA}))$
601 CONTINUE

GOTO 4000


```
    700 DO 701 I = FROM(M),UPTO(M)
            THETA = THETA + (PI/180.)
            GAMMA = THETA/BETA
            S(I) = S(C) + H(M)*(GAMMA - (SIN(PI*GAMMA)/PI))
            V(I) = (H(M)/BETA)*( 1 - COS (PI*GAMMA))
    7 0 1 ~ C O N T I N U E ~
    GOTO 4000
```

\%
800 DO 801 I = FROM(M),UPTO(M)
THETA $=$ THETA $+($ PI $/ 180$.
GAMMA $=$ THETA/BETA
$S(I)=S(C)+H(M) *($ GAMMA $+(S I N(P I * G A M M A) / P I))$
$V(I)=(H(M) / B E T A) *(1+\operatorname{COS}(P I * G A M M A))$
801 CONTINUE
GOTO 4000
N:
900 DO $901 \mathrm{I}=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+($ PI/180. $)$
GAMMA $=$ THETA/BETA
$S(I)=-H(M) *(1-G A M M A+(S I N(P I * G A M M A) / P I))$
$V(I)=(H(M) / B E T A) *(1-\operatorname{COS}(P I * G A M M A))$
901 CONTINUE
GOTO 4000

1000 DO $1001 \mathrm{I}=\mathrm{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+(\mathrm{PI} / 180$.
GAMMA $=$ THETA/BETA
$S(I)=-H(M) *(1-G A M M A-(S I N(P I * G A M M A) / P I))$
$V(I)=(H(M) / B E T A) \div(1+\operatorname{COS}(P I \div G A M M A))$
1001 CONTINUE
GOTO 4000

1100 DO $1101 \mathrm{I}=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+(P I / 180$.
GAMMA $=$ THETA/BETA
$S(I)=S(C)+H(M) *(G A M M A-(S I N(2 * P I * G A M M A) /(2 * P I)))$

```
    V(I)=(H(M)/BETA)*( 1-COS(2*PI*GAMMA))
1101 CONTINUE
GOTO 4000
```



```
1200 DO 1201 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    S(I) = - H(M)*( 1 - GAMMA + (SIN (2*PI*GAMMA )/(2*PI)))
    V(I) = (H(M)/BETA)*( 1 - COS (2*PI*GAMMA ))
1201 CONTINUE
GOTO 4000
```


1300 DO $1301 \mathrm{I}=\mathrm{FROM}(\mathrm{M})$, UPTO (M)
THETA $=$ THETA $+(\mathrm{PI} / 180$.
$S(I)=S(C)+(H(M) / B E T A) * T H E T A$
$V(I)=H(M) / B E T A$
1301 CONTINUE
GOTO 4000

1400 DO $1401 \mathrm{I}=\mathrm{FROM}(\mathrm{M})$, UPTO (M)
THETA $=\mathrm{THETA}+(\mathrm{PI} / 180$.
GAMMA $=$ THETA/BETA
IF (GAMMA .LE 0.5 ) THEN
$S(I)=S(C)+2 * H(M) *($ GAMMA $* * 2)$
$V(I)=4 \div H(M) \div$ GAMMA/BETA
ELSE
$S(I)=S(C)+H(M)-2 \div H(M) *((1-G A M M A) * * 2)$
$V(I)=4 * H(M) *(1-G A M M A) / B E T A$
ENDIF

1401 CONTINUE

GOTO 4000
\%
1500 DO 1501 I $=$ FROM (M), UPTO(M)
THETA $=$ THETA $+(\mathrm{PI} / 180$.
$S(I)=S(C)+3 *($ THETA**2) - $2 \div($ THETA $* \div 3)$
$V(I)=6 \%$ THETA $\%(1-$ THETA $)$
1501 CONTINUE
GOTO 4000

* 1600 DO $1601 \mathrm{I}=\mathrm{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$

THETA $=$ THETA $+(\mathrm{PI} / 180$.
$S(I)=S(C)+10 \div($ THETA $* * 3)-15 *($ THETA $\div \div 4)+6 \div($ THETA**5 $)$
$V(I)=30 *($ THETA $* \div 2)-60 \div($ THETA** $* 3)+30 \div($ THETA $\div \div 4)$
1601 CONTINUE

GOTO 4000


```
1700 DO 1701 I = FROM(M),UPTO(M)
            THETA = THETA + (PI/180.)
                        S(I) =S (C)+35*(THETA**4)-84*(THETA**5)+70*(THETA**6)
            非 -20%(THETA***7)
            V(I) = 140*(THETA**3)-420*(THETA**4)+420* (THETA**5 )
    揓 -140%(THETA***6)
1701 CONTINUE
```

GOTO 4000

1800 DO $1801 \mathrm{I}=\mathrm{FROM}(\mathrm{M})$, UPTO (M)
THETA $=$ THETA $+(\mathrm{PI} / 180$.
GAMMA $=$ THETA/BETA
$S(I)=S(C)+H(M) \div($ GAMMA $-15 \div(S I N(2 \div P I * G A M M A)) /(32 \div P I)$
\# $\quad-(\mathrm{SIN}(6 \div \mathrm{PI} \div \mathrm{GAMMA})) /(96 \div \mathrm{PI}))$
$V(I)=(H(M) / B E T A) *(1-(15 / 16) \div \operatorname{COS}(2 \div P I * G A M M A)$
\# $\quad-\operatorname{COS}(6 \div \mathrm{PI} \div$ GAMMA $) / 16)$

1801 CONTINUE
GOTO 4000

1900 DO $1901 \mathrm{I}=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})$
THETA $=$ THETA $+(P I / 180$.
GAMMA $=$ THETA/BETA
$S(I)=S(C)+H(M) \div G A M M A-(H(M) /(2 \div P I)) \div(27 \div S I N(2 \div P I \div G A M M A) / 28$
$\# \quad+\mathrm{SIN}(6 \div \mathrm{PI} \div \mathrm{GAMMA}) / 84)$
$V(I)=(H(M) / B E T A) \div(1-27 \div \operatorname{COS}(2 * P I * G A M M A) / 28-C O S(6 \div P I * G A M M A) / 28)$
1901 CONTINUE
GOTO 4000
\#; If motion is Freudenstein 1-3-5 Harmonic then go to this section. $2000 \mathrm{~B}=1125 . / 1192$.

```
    DO 2001 I = FROM(M),UPTO(M)
            THETA = THETA + (PI/180.)
            GAMMA = THETA/BETA
            S(I) = S (C) + H(M)*GAMMA - (H(M)*B/(2*PI))*(SIN(2*PI*GAMMA)
    # + SIN(6*PI*GAMMA)/54 + SIN(10*PI*GAMMA)/1250)
    V(I) = (H(M)/BETA )*(1-B*(COS (2*PI*GAMMA )+COS (6*PI*GAMMA )/18
            + COS(10*PI*GAMMA)/250))
```

2001 CONTINUE

GOTO 4000


```
2100 DO 2101 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
```

    IF (GAMMA .LT. 1/8) THEN
        \(S(I)=S(C)+H(M) *(0.43989 * G A M M A-0.035014 * S I N(4 * P I * G A M M A))\)
        \(V(I)=(0.43989 * H(M) / B E T A) *(1-\operatorname{COS}(4 * P I * G A M M A))\)
    ELSE
        IF (GAMMA .LT. 7/8) THEN
            \(\mathrm{S}(\mathrm{I})=\mathrm{S}(\mathrm{C})+\mathrm{H}(\mathrm{M}) *(0.28005+0.43989 *\) GAMMA
    \# $\quad-0.315055 * \operatorname{COS}((4 * P I / 3) * G A M M A-P I / 6))$
$\mathrm{V}(\mathrm{I})=(\mathrm{H}(\mathrm{M}) / \mathrm{BETA}) *(0.43989+1.31967 * \operatorname{SIN}((4 * \mathrm{PI} / 3) * \mathrm{GAMMA}-\mathrm{PI} / 6))$
ELSE
$\mathrm{S}(\mathrm{I})=\mathrm{S}(\mathrm{C})+\mathrm{H}(\mathrm{M}) *(0.56010+0.43989 *$ GAMMA
- $0.035014 * \operatorname{SIN}(2 * P I *(2 *$ GAMMA - 1$)))$
$\mathrm{V}(\mathrm{I})=(\mathrm{H}(\mathrm{M}) / \mathrm{BETA}) *(0.43989 *(1-\operatorname{COS}(2 * \mathrm{PI} *(2 *$ GAMMA -1$))))$
ENDIF

## ENDIF

## 2101 CONTINUE

GOTO 4000


```
2200 DO 2201 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
```

    IF (GAMMA .LT. 1/8) GOTO 2210
    IF (GAMMA .LT. 3/8) GOTO 2220
    IF (GAMMA .LT. 1/2) GOTO 2230
    IF (GAMMA .LT. 5/8) GOTO 2240
    IF (GAMMA .LT. 7/8) GOTO 2250
    \(\mathrm{S}(\mathrm{I})=\mathrm{S}(\mathrm{C})+\mathrm{H}(\mathrm{M}) *(0.6110154+0.3889845 *\) GAMMA
    \# $\quad+0.0309544 \div$ SIN(PI*(4*GAMMA-3)))
$\mathrm{V}(\mathrm{I})=(\mathrm{H}(\mathrm{M}) / \mathrm{BETA}) *(0.3889845+0.3889845 * \operatorname{COS}(\mathrm{PI} *(4 *$ GAMMA-3))$)$

GOTO 2201

```
2210 S(I) = S(C) + 0.09724612*H(M)*(4*GAMMA - SIN(4*PI*GAMMA)/PI)
    V(I) = (0.3889845*H(M)/BETA)*(1 - COS(4*PI*GAMMA))
    GOTO 2201
2220 S(I) = S (C) + H(M)*(2.44406184*(GAMMA**2) - 0.22203097*GAMMA
    # + 0.00723407)
        V(I) = (H(M)/BETA)*(4.888124*GAMMA - 0.222031)
        GOTO 2201
2230 S(I) = S (C) + H(M)*(1.6110154*GAMMA-0.0309544*SIN(PI*(4*GAMMA-1))
    # - 0.3055077)
        V(I) = (H(M)/BETA)*(1.6110154 - 0.3889845*COS(PI*(4*GAMMA - 1)))
        GOTO 2201
2240 S(I) = S(C) + H(M)*(1.6110154*GAMMA+0.0309544*SIN(PI*(4*GAMMA-2))
    # - 0.3055077)
        V(I) = (H(M)/BETA)*(1.6110154 + 0.3889845*COS(PI*(4*GAMMA - 2)))
        GOTO 2201
2250 S(I) = S(C) + H(M)*(4.6660917*GAMMA - 2.44406184*(GAMMA**2)
    # - 1.2292648)
        V(I) = (H(M)/BETA)*(4.6660917 - 4.888124*GGAMMA)
2201 CONTINUE
4000 CONTINUE
```

    RETURN
    END
    

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六－THE FOLLOWER AT ONE DEGREE INTERVALS．THE SUBROUTINE IS 六市
다 STRUCTURED SUCH THAT THE CALCULATIONS FOR EACH SEGMENT ARE＊
DONE INSIDE A DO LOOP．THIS SUBROUTINE IS NOT REQUIRED IN N
产 ALL THE DISPLAY OPTIONS CHOSEN BY THE USER．HOWEVER IT MUST 京安
为 $\operatorname{te}$ CALLED IN CASE THE USER DECIDES TO DOCUMENT THE DATA．六交
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SUBROUTINE CURVES

COMMON／CAM1／TYPE，BASE，RPM，WIDTH，DESI，ROLL，ECC，RA，RF COMMON／COUNT／I，N，M，B，PI，C，R，L

```
    COMMON/ANGLE/ BETA, THETA, GAMMA
    COMMON/MTN/ E(25), H(20), FROM(20), UPTO(20)
    COMMON/CRV/ S(360), V(360), A(360), J(360)
    COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
    COMMON/DISP/ OUT, MONI
    INTEGER I, N, M, C, R
    INTEGER TYPE, OUT, DESI, MONI
    INTEGER E, FROM, UPTO
    REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
    REAL ROLL, ECC, RA, RF
    REAL S, V, A, J
    REAL ALPHA, X, Y, Z, H
    DO 4000 M= 1,N
    BETA = (UPTO(M)- UPTO(M-1))}\divPI/180.
    C = UPTO(M-1)
    THETA = 0.
    GO TO (100,200, 300,400,500,600,700,800,900,1000,1100, 1200,1300,
#
    1400, 1500, 1600, 1700, 1800, 1900, 2000, 2100, 2200), E (M)
```



```
    DO 11 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = 0.
    J(I) = 0.
    11 CONTINUE
    GOTO 4000
```



```
    100 DO 101 I = FROM(M),UPTO(M)
            THETA = THETA + (PI/180.)
            GAMMA = THETA/BETA
            A(I) = H(M)*((PI*0.5/BETA )**2)*(COS (PI*GAMMA/2))
            J(I) = -H(M)*((PI*0.5/BETA )**3)*(SIN (PI*GAMMA/2))
    101 CONTINUE
        GOTO 4000
```



```
    200 DO 201 I = FROM(M),UPTO(M)
        THETA = THETA + (PI/180.)
        GAMMA = THETA/BETA
        A(I) = -((PI/(2*BETA))**2)*H(M)*(SIN(PI*GAMMA/2))
        J(I) = -H(M)*((PI*0.5/BETA )**3)*(COS(PI*GAMMA/2))
    201 CONTINUE
    GOTO 4000
```



```
300 DO 301 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = ((PI/ (2*BETA ) )**2)*H(M)*(COS (PI*GAMMA / ) )
    J(I) = -H(M)*((PI*0.5/BETA )**3)*(SIN(PI*GAMMA/2))
301 CONTINUE
    GOTO 4000
```



```
400 DO 401 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = -((PI/(2*BETA))**2)*H(M)*(SIN(PI*GAMMA / 2))
    J(I) = - H(M)*((PI*0.5/BETA )**3)*(COS (PI*GAMMA/2))
401 CONTINUE
    GOTO 4000
W\% Wh:
    500 DO 501 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = 2*((PI/(2*BETA ) )**2)*H(M)*(COS (PI*GAMMA ))
    J(I) = -4*((PI/(2*BETA ))**3)*H(M)*(SIN (PI*GAMMA))
501 CONTINUE
    GOTO 4000
```



```
    600 DO 601 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = 2*((PI/(2*BETA ) )**2)*H(M)*(COS (PI*GAMMA))
    J(I) = -4*((PI/ (2*BETA ))**3)*H(M)*(SIN(PI*GAMMA))
    6 0 1 ~ C O N T I N U E ~
    GOTO 4000
```



```
    700 DO 701 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = (PI/BETA )*(H(M)/BETA )*(SIN (PI*GAMMA )
    J(I) = ((PI/BETA )**2)*(H(M)/BETA )*(COS (PI*GAMMA))
    701 CONTINUE
    GOTO 4000
```

    800 DO 801 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = - (PI/BETA)*(H(M)/BETA)*(SIN(PI*GAMMA))
    J(I) = - ((PI/BETA )**2)*(H(M)/BETA )*(COS (PI*GAMMA))
    801 CONTINUE
    GOTO 4000
    ```

```

    900 DO 901 I = FROM(M),UPTO(M)
        THETA = THETA + (PI/180.)
        GAMMA = THETA/BETA
        A(I) = (PI/BETA)*(H(M)/BETA)*(SIN(PI*GAMMA))
        J(I) =((PI/BETA)**2)*(H(M)/BETA)*(COS (PI*GAMMA))
    901 CONTINUE
    GOTO 4000
    ```

```

    1000 DO 1001 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = -(PI/BETA )*(H(M)/BETA)*(SIN(PI*GAMMA))
    J(I) = -((PI/BETA )**2)*(H(M)/BETA)*(COS (PI*GAMMA))
    1001 CONTINUE
GOTO 4000

```

```

    1100 DO 1101 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = 2*(PI/BETA)*(H(M)/BETA)*(SIN(2*PI*GAMMA))
    J(I) = ((2*(PI/BETA))**2)*(H(M)/BETA)*(COS (2*PI*GAMMA))
    1101 CONTINUE
    GOTO 4000
    ```

```

    1200 DO 1201 I = FROM(M),UPTO(M)
    THETA = THETA + (PI/180.)
    GAMMA = THETA/BETA
    A(I) = 2*(PI/BETA)*(H(M)/BETA)*(SIN(2*PI*GAMMA))
    J(I) = ((2*(PI/BETA ))**2)*(H(M)/BETA)*(COS (2*PI*GAMMA))
    1201 CONTINUE
    GOTO 4000
    ```


1300 DO 1301 I \(=\operatorname{FROM}(M), U P T O(M)\)
```

        A(I) = 0
        J(I) = 0.
    1301 CONTINUE
GOTO 4000

```

```

1400 DO 1401 I = FROM(M),UPTO(M)
THETA = THETA + (PI/180.)
GAMMA = THETA/BETA
IF (GAMMA .LE. O.5) THEN
A(I) = 4*H(M)/(BETA**2)
ELSE
A(I) = -4*H(M)/(BETA**2)
ENDIF
J(I) = 0.
1401 CONTINUE
GOTO 4000

```

1500 DO \(1501 \mathrm{I}=\mathrm{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})\)
    THETA \(=\) THETA \(+(\) PI/180. \()\)
    \(A(I)=6 *(1-2 *\) THETA \()\)
    \(J(I)=-12\).
1501 CONTINUE
    GOTO 4000

```

1600 DO 1601 I = FROM(M),UPTO(M)

```
    THETA \(=\) THETA \(+(\) PI \(/ 180\).
    \(A(I)=60 *(\) THETA \()-180 \%(\) THETA \(* * 2)+120 \%(\) THETA \(* * 3)\)
    \(J(I)=60 .-360 \%(\) THETA \()+360 \%(\) THETA \(* 2)\)
1601 CONTINUE
    GOTO 4000

```

1700 DO 1701 I = FROM(M),UPTO(M)
THETA = THETA + (PI/180.)
A(I) = 420*(THETA**2) - 1680*(THETA**3) + 2100*(THETA**4)
\# - 840*(THETA**5)
J(I) = 840*(THETA) - 5040*(THETA**2) + 8400*(THETA**3)
\# - 4200*(THETA**4)

```
    1701 CONTINUE
    GOTO 4000

```

    1 8 0 0 ~ D O ~ 1 8 0 1 ~ I ~ = ~ F R O M ( M ) , U P T O ( M )
    THETA = THETA + (PI/180.)
    ```
```

        GAMMA = THETA/BETA
        A(I)=(PI/8)*H(M)/(BETA**2)*(15*SIN(2*PI*GAMMA)
            +3*SIN(6*PI*GAMMA))
            J(I) = ((PI**2)/4)*(H(M)/(BETA**3))*( 15*COS (2*PI*GAMMA)
                + 9*COS (6%PI*GAMMA))
    1801 CONTINUE
GOTO 4000

```

1900 DO \(1901 \mathrm{I}=\operatorname{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})\)
            THETA \(=\) THETA \(+(P I / 180\).
            GAMMA \(=\) THETA/BETA
            \(A(I)=(2 * P I / B E T A) *(H(M) / B E T A) *(27 * S I N(2 \div P I * G A M M A) / 28\)
    \# + \(\quad 3 \div \operatorname{SIN}(6 \div P I \div G A M M A) / 28)\)
    \(J(I)=((2 \div\) PI \(/\) BETA \() \div 2) \div(H(M) /\) BETA \() \div(27 \div \operatorname{COS}(2 \div P I \div\) GAMMA \() / 28\)
    \# \(\quad+9 \div \operatorname{COS}(6 \div \mathrm{PI} \div\) GAMMA \() / 28)\)
1901 CONTINUE
    GOTO 4000
찬 If motion is Freudenstein 1-3-5 Harmonic then go to this section.
\(2000 \mathrm{~B}=1125 . / 1192\).
    DO \(2001 \mathrm{I}=\mathrm{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})\)
            THETA \(=\) THETA \(+(\mathrm{PI} / 180\).
            GAMMA = THETA/BETA
            \(A(I)=(2 * P I / B E T A) \div(H(M) / B E T A) \div B *(S I N(2 * P I \div G A M M A)\)
    \# \(+\operatorname{SIN}(6 \div P I \div G A M M A) / 6+\operatorname{SIN}(10 \div P I * G A M M A) / 50)\)
            \(J(I)=((2 \div P I / B E T A) * \div 2) \div(H(M) / B E T A) * B *(C O S(2 \div P I \div G A M M A)\)
    \# \(\quad+\operatorname{COS}(6 * P I * G A M M A) / 2+\operatorname{COS}(10 * P I * G A M M A) / 10)\)
2001 CONTINUE
    GOTO 4000
K
2100
    DO \(2101 \mathrm{I}=\mathrm{FROM}(\mathrm{M}), \mathrm{UPTO}(\mathrm{M})\)
    THETA \(=\) THETA \(+(P I / 180\).
    GAMMA \(=\) THETA/BETA
    IF (GAMMA .LT. 1/8) THEN
        \(A(I)=5.52794 *(H(M) /(B E T A * \div 2)) * S I N(4 * P I * G A M M A)\)
        \(J(I)=69.4659 *(H(M) /(B E T A * * 3)) \div \operatorname{COS}(4 \div \mathrm{PI} * G A M M A)\)
    ELSE
        IF (GAMMA .LT. 7/8) THEN
            \(A(I)=5.52794 \div(\mathrm{H}(M) /(\mathrm{BETA} \div \div 2)) \div \operatorname{COS}((4 \div \mathrm{PI} / 3) \div \mathrm{GAMMA}-\mathrm{PI} / 6)\)
            \(J(I)=-23.1553 *(H(M) /(B E T A * \div 3)) * S I N((4 * P I / 3) \div\) GAMMA-PI \(/ 6)\)
        ELSE
            \(A(I)=5.52794 \div(H(M) /(B E T A * * 2)) \div \operatorname{SIN}(2 \div P I \div(2 * G A M M A-1))\)
            \(J(I)=69.4659 *(H(M) /(B E T A * * 3)) * \operatorname{COS}(2 * P I *(2 * G A M M A-1))\)
        ENDIF
    ENDIF
```

2200 DO 2201 I = FROM(M),UPTO(M)
THETA = THETA + (PI/180.)
GAMMA = THETA/BETA
IF (GAMMA .LT. 1/8) GOTO 2210
IF (GAMMA .LT. 3/8) GOTO 2220
IF (GAMMA .LT. 1/2) GOTO 2230
IF (GAMMA .LT. 5/8) GOTO 2240
IF (GAMMA .LT. 7/8) GOTO 2250
A(I) = -4.888124*(H(M)/(BETA**2))*SIN(PI*(4*GAMMA-3))
J(I) = -61.425769*(H(M)/(BETA**3))*COS(PI*(4*GAMMA-3))
GOTO 2201
2210 A(I) = 4.888124%(H(M)/(BETA**2))*SIN(4*PI*GAMMA)
J(I) = 61.425769*(H(M)/(BETA**3))*COS (4*PI*GAMMA)
GOTO 2201
2220 A(I) = (H(M)/(BETA穴2))*(4.888124)
J(I) = 0.
GOTO 2201
2230 A(I) = 4.888124*(H(M)/(BETA**2))*SIN(PI*(4*GAMMA - 1))
J(I) = 61.425769*(H(M)/(BETA**3))*COS(PI*(4*GAMMA - 1))
GOTO 2201
2240 A(I) = -4.888124*(H(M)/(BETA**2))*SIN(PI*(4*GAMMA - 2))
J(I) = -61.425769*(H(M)/(BETA**3))*COS(PI*(4*GAMMA - 2))
GOTO 2201
2250 A(I) = -4.888124*(H(M)/(BETA**2 ) )
J(I) = 0.

```

2201 CONTINUE

\section*{4000 CONTINUE}

RETURN
END
\begin{tabular}{|c|c|}
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\hline & \\
\hline 京京 & SUBROUTINE CHANGE \\
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\end{tabular}


\section*{SUBROUTINE CHANGE}

COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF
COMMON/COUNT/ I, N, M, B, PI, C, R, L
COMMON/ANGLE/ BETA, THETA, GAMMA
COMMON/MTN/ E(25), H(20), FROM(20), UPTO (20)
COMMON/CRV/ S(360), V(360), A(360), J(360)
COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
COMMON/DISP/ OUT, MONI
INTEGER I, N, M, C, R
INTEGER TYPE, OUT, DESI, MONI
INTEGER E, FROM, UPTO
REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
REAL ROLL, ECC, RA, RF
REAL S, V, A, J
REAL ALPHA, X, Y, Z, H
CHARACTER CHAR
10 FORMAT(A1)
\(100 \operatorname{WRITE}(6, *)\) ', Please select option'
\(\operatorname{WRITE}(6, *)\)
WRITE (6,*)
\(\operatorname{WRITE}(6, *)\) ' \(1:\) Follower Type'
WRITE \((6, *)\) 2: Basic Data'
\(\operatorname{WRITE}(6, *)\) ' 3: Output Type'
\(\operatorname{WRITE}(6, *)\) ' 4: Design Req.'
WRITE \((6, *)\) 5: Motion Data'
WRITE \((6, *)\) ' 6: NO MORE CHANGES'
\(\operatorname{READ}(5, *) R\)
IF (R .LT. 1 .OR. R .GT. 6) THEN
WRITE \((6, *)\) '
ERROR!! INVALID SELECTION!'
GOTO 100
ENDIF
GO TO \((200,300,400,500,600,700), R\)
```

200 CALL FOLLOW
GOTO 100
300 CALL BASIC
GOTO 100
400 CALL OUTPUT
GOTO 100
500 CALL DESIGN
GOTO 100
6 0 0 ~ C A L L ~ M O T I O N
GOTO 100
700 RETURN
END

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安穴 交安



六市
的六 THIS SUBROUTINE COMPUTES THE X AND Y COORDINATES OF THE CAM 相
추 PROFILE．IT USES THE DATA MADE AVAILABLE BY THE USER FOR 玄
\(\%\) THE CAM AND THE FOLLOWER．THE DATA FOR MOTION CHARACTERISTICS 六
六市 MAY ALSO BE USED IN SOME CASES．THIS SUBROUTINE IS NOT REQUIRED 交
安：FOR ALL DISPLAY OPTIONS CHOSEN BY THE USER．HOWEVER IT MUST 六市
\(\therefore \quad\) BE CALLED IN CASE THE USER DECIDES TO DOCUMENT THE DATA．
末宋
篗

\section*{SUBROUTINE PROFIL}

COMMON／CAM1／TYPE，BASE，RPM，WIDTH，DESI，ROLL，ECC，RA，RF
COMMON／COUNT／I，N，M，B，PI，C，R，L
COMMON／ANGLE／BETA，THETA，GAMMA
COMMON／MTN／E（25），H（20），FROM（20），UPTO（20）
COMMON／CRV／S \((360), V(360), A(360), J(360)\)
COMMON／COOD／ALPHA（360），X（361），Y（361），Z（361）
COMMON／DISP／OUT，MONI
INTEGER I，N，M，C，R
```

    INTEGER TYPE, OUT, DESI, MONI
    INTEGER E, FROM, UPTO
    REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
    REAL ROLL, ECC, RA, RF
    REAL S, V, A, J
    REAL ALPHA, X, Y, Z, H
    GO TO (100,200,300,400,500,600,700),TYPE
    ```
```

100 DO 101 I = 1, 360

```
100 DO 101 I = 1, 360
    THETA = I*PI/180.
    THETA = I*PI/180.
    X(I) = (BASE + S(I))*COS(THETA)
    X(I) = (BASE + S(I))*COS(THETA)
    Y(I) = (BASE + S(I))*SIN(THETA)
    Y(I) = (BASE + S(I))*SIN(THETA)
101 CONTINUE
    GO TO 1000
200 DO 201 I = 1, 360
    THETA = I*PI/180.
    ALPHA(I) = ATAN(V(I)/(BASE+ROLL+S(I)))
    X(I) = (BASE+ROLL+S(I))*COS(THETA) - ROLL*COS(THETA-ALPHA(I))
    Y(I) = (BASE+ROLL+S(I))*SIN(THETA) - ROLL*SIN(THETA-ALPHA(I))
201 CONTINUE
    GO TO 1000
300 DO 301 I = 1, 360
    THETA = I }*\textrm{PI}/180
    ALPHA(I) = ATAN((V(I)-ECC)/(BASE+ROLL+S(I)))
    X(I) = ((((BASE+ROLL)**2-(ECC)**2)**0.5)+S(I))*COS (THETA)
                                - ROLL*COS(THETA-ALPHA(I)) + ECC*SIN(THETA)
    Y(I) = ((((BASE+ROLL)**2-(ECC)**2)**0.5)+S(I))*SIN(THETA)
        - ROLL*SIN(THETA-ALPHA(I)) - ECC*COS(THETA)
301 CONTINUE
    GO TO 1000
400 DO 401 I = 1, 360
    THETA = I*PI/180.
    PHIZ = ACOS (((RF**2) +(RA**2) -((BASE+ROLL)**2)) / (2*RA*RF))
    ALPHA(I)=ATAN(1/TAN(S(I)+PHIZ)-(RF*(1-V(I))/(RA*SIN(S (I)+PHIZ))))
        X(I) = RA*COS(THETA) - RF*COS(THETA - S(I) - PHIZ)
    # - ROLL*SIN(ALPHA(I) - THETA + S(I) + PHIZ)
        Y(I) = RA*SIN(THETA) - RF*SIN(THETA - S(I) - PHIZ)
    # - ROLL*COS(ALPHA(I) - THETA + S(I) + PHIZ)
4 0 1 ~ C O N T I N U E ~
    GO TO 1000
500 DO 501 I = 1, 360
    THETA = I*PI/180.
    X(I) = (BASE+S(I))*COS(THETA) - V(I)*SIN(THETA)
    Y(I) = (BASE+S(I))*SIN(THETA) + V(I)*COS(THETA)
```

```
    501 CONTINUE
        GO TO 1000
    600 DO 601 I = 1, 360
        THETA = I *PI/180.
        PHIZ = ASIN(BASE/RA)
        DUMB = PHIZ + S(I) - THETA
        DUMA = (COS (THETA + DUMB))/(V(I) - 1)
        X(I) = RA*(COS(THETA) + DUMA*COS(DUMB))
        Y(I) = RA*(SIN(THETA) - DUMA*SIN(DUMB))
    6 0 1 ~ C O N T I N U E ~
        GO TO 1000
    700 DO 701 I = 1, 360
        THETA = I*PI/180.
        PHIZ = ASIN((ECC+BASE)/RA)
        DUMB = PHIZ + S(I) - THETA
        DUMA = (COS (THETA + DUMB))/(V(I) - 1)
        X(I) = RA*(COS(THETA) + DUMA*COS(DUMB)) + ECC*SIN(DUMB)
        Y(I) = RA*(SIN(THETA) - DUMA*SIN(DUMB)) + ECC*COS(DUMB)
    7 0 1 ~ C O N T I N U E ~
1000 RETURN
        END
```


## 

重大宋

大六 SUBROUTINE GKS 哭
＊大 ㄷ

K高央
다 THIS SUBROUTINE DISPLAYS THE OUTPUT PARAMETERS OF THE CAM TO N
눈 THE SCREEN IN FORM OF 2－D GRAPHICS．IT ALLOWS THE USER TO 六
＊SIMULATE THE MOTION OF THE CAM．THE SIMUILATION SMOOTHNESS＊
字 DEPENDS ON THE DEGREE STEPS SELECTED BY THE USER．市完
THIS ROUTINE IS CALLED ONLY IF THE DISPLAY SYSTEM CHOSEN BY 安 T
安 THE USER AT THE BEGINING OF THE PROGRAM IS GKS．安
六六 东笑

｜

SUBROUTINE GKS
COMMON／CAM1／TYPE，BASE，RPM，WIDTH，DESI，ROLL，ECC，RA，RF

```
    COMMON/COUNT/ I, N, M, B, PI, C, R, L
    COMMON/ANGLE/ BETA, THETA, GAMMA
    COMMON/MTN/ E(25), H(20), FROM(20), UPTO(20)
    COMMON/CRV/ S(360),V(360), A(360), J(360)
    COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
    COMMON/DISP/ OUT, MONI
    INTEGER I, N, M, C, R, W
    INTEGER TYPE, OUT, DESI, MONI
    INTEGER E, FROM, UPTO
    REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
    REAL ROLL, ECC, RA, RF
    REAL S, V, A, J
    REAL ALPHA, X, Y, Z, H
    REAL XFA(4), YFA(4)
    REAL NORMS (360), NORMV (360), NORMA(360), NORMJ (360), NORMSG(360)
    REAL XS(361), YS(361), XZ(361)
    REAL XA(361), YA(361), ZA(361)
    REAL XV(361), YV(361), ZV(361)
    REAL XJ(361), YJ(361), ZJ(361)
    REAL XAXIS(5), YAXIS(5)
    CHARACTER*1 CHAR
    CHARACTER\div8 TYP
    CHARACTER:%8 DIS 1
    CHARACTER\div8 VEL1
    CHARACTER\div8 ACC1
    CHARACTER*8 JER1
    CHARACTER*8 DIS2
    CHARACTER\div8 VEL2
    CHARACTER*8 ACC2
    CHARACTER*% JER2
    10 FORMAT(A1)
    L = L + 1
    IF (L .LT. 2) THEN
    CALL UOPNSF(1, 'ERROR DATA',1,JERROR)
        IF (JERROR .NE. 0) THEN
        WRITE (*,*)'CANNOT OPEN ERROR LOGGING FILE ',JERROR
        STOP
        END IF
    END IF
    KERFIL = 1
```



```
    OPEN (UNIT = 20, FILE = 'ERRFIL')

CALL GOPKS（20）
\％ IF（L ．LT．2）THEN
100 WRITE（5，＊）＇PLEASE ENTER TERMINAL TYPE＇ READ（5，FMT＝200）KWKTYP
＊
CALL GQEWK（ \(0, \mathrm{KERROR}, \mathrm{KNUM}, \mathrm{KJUNK}\) ）
TYP＝＇INVALID＇
DO \(300 \mathrm{I}=1\) ，KNUM
\(\mathrm{W}=\mathrm{I}\)
CALL GQEWK（W，KERROR，KJUNK，KTYPE）
IF（KWKTYP ．EQ．KTYPE）TYP＝＇VALID＇
300 CONTINUE
IF（TYP ．EQ．＇INVALID＇）GOTO 100
END IF
\＃：＊：＊Open and activate workstation 1，use LUN 5，and KWKTYP for type．N：＊＊
CALL GOPWK（ \(1,5, \mathrm{KW}\) KTYP）
CALL GACWK（1）

DUMMY1 \(=.75 / 4\)
DUMMY2 \(=.75 / 2\)
DUMMY3 \(=3 \% .75 / 4\)



SCALE \(1=0.001\)
DO \(350 \mathrm{I}=1,360\)
SCALE 1 ＝MAX（ ABS（X（I）），SCALE1）
350 CONTINUE
DO \(375 \mathrm{I}=1,360\)
SCALE \(1=\operatorname{MAX}(\operatorname{ABS}(Y(I))\) ，SCALE1）
375 CONTINUE
SCALE \(1=2 *\) SCALE 1
\％
六六办
SCALE2 \(=0.001\)
DO \(400 \mathrm{I}=1,360\)
SCALE2 \(=\operatorname{MAX}(\operatorname{ABS}(\mathrm{S}(\mathrm{I}))\), SCALE2）
```

400 CONTINUE
DO 450 I=1,360
NORMS(I)= S(I)/SCALE2
450 CONTINUE

```



SCALE3 \(=0.001\)
DO \(500 \mathrm{I}=1,360\)
SCALE3 \(=\operatorname{MAX}(\operatorname{ABS}(V(I)), \operatorname{SCALE} 3)\)
500 CONTINUE
DO \(550 \mathrm{I}=1,360\)
\(\operatorname{NORMV}(I)=V(I) /\) SCALE3
550 CONTINUE
 \%

SCALE4 \(=0.001\)
DO \(600 \mathrm{I}=1,360\)
SCALE4 \(=\) MAX ( ABS (A (I)), SCALE4)
600 CONTINUE
DO \(650 \mathrm{I}=1,360\)
NORMA (I) \(=A(I) / S_{\text {CALE }} 4\)
650 CONTINUE




SCALE5 \(=0.001\)
DO \(700 \mathrm{I}=1,360\)
SCALE5 \(=\) MAX \((\operatorname{ABS}(J(I))\), SCALE5 \()\)
700 CONTINUE
DO \(650 \mathrm{I}=1,360\)
\(\operatorname{NORMJ}(I)=J(I) /\) SCALE 5
750 CONTINUE

K


SCAL2 \(=\) SCALE2 +0.5
SCAL3 \(=\) SCALE3 +0.5
SCAL4 \(=\) SCALE4 +0.5
SCAL5 \(=\) SCALE5 +0.5

ISCAL2 \(=\) INT(SCAL2)
ISCAL3 \(=\) INT (SCAL3)
ISCAL4 \(=\) INT(SCAL4)
ISCAL5 \(=\) INT (SCAL5)
WRITE (DIS1, '(I6)') ISCAL2
```

WRITE(VEL1,'(I6)') ISCAL3
WRITE(ACC1,'(I6)') ISCAL4

- WRITE(JER1,'(I6)') ISCAL5
WRITE(DIS2,'(I6)') -ISCAL2
WRITE(VEL2,'(I6)') -ISCAL3
WRITE(ACC2,'(I6)') -ISCAL4
WRITE(JER2,'(I6)') -ISCAL5

```

\(\operatorname{CALL} \operatorname{GSWN}(1,0 ., 1 ., 0 ., 1\).
CALL GSWN (2,-SCALE1,SCALE1,-SCALE1,SCALE1)
CALL GSVP \((2,0 ., .75,0 ., .75)\)
CALL GSWN ( \(3,0 ., 1 ., 0 ., 1\). )
CALL GSVP ( \(3, .75,1 ., 0 .\), DUMMY1)
CALL GSWN (4,0.,1.,0.,1.)
CALL GSVP (4,.75,1.,DUMMY1,DUMMY2)
CALL GSWN (5,0., 1., 0., 1.)
CALL GSVP (5,.75,1., DUMMY2,DUMMY3)
CALL GSWN (6,0.,1.,0., 1.)
CALL GSVP ( \(6, .75,1 .\), DUMMY \(3, .75\) )
CALL GSWN (7,0.,1.,0.,1.)
CALL GSVP ( \(7,0 ., .5, .375, .75\) )
CALL GSWN (8,0.,1.,0.,1.)
CALL GSVP \((8, .5,1 ., .375, .75)\)
CALL GSWN (9,0.,1.,0.,1.)
CALL GSVP (9,.5,1.,0.,.375)
CALL GSWN (10,0.,1.,0.,1.)
CALL GSVP ( \(10,0 ., .5,0 ., .375\) )

\section*{CALL GSELNT (1)}
.

CALL GQMDS (KWKTYP, KERROR,XUNITS, XMETER,YMETER,XRAST,YRAST) IF (KERROR .NE. 0) WRITE (5, *) 'INQUIRE SIZE ERROR'

度
 values to the correct units.
```

    IF (XUNITS .EQ. O) THEN
    XMAX=XMETER
    YMAX=YMETER
    ELSE
    XMAX=XRAST
    YMAX=YRAST
    ENDIF
    CALL GSWKWN (1,0.,1.,0.,.75)
    CALL GSWKVP (1,0.,XMAX,0.,YMAX)
    CALL GUWK (1,1)
    ```

```

CALL GSELNT (2)
XFA(1) = -SCALE1
XFA(2) = SCALE1
XFA(3) = SCALE1
XFA(4) = -SCALE1
YFA(1) = -SCALE1
YFA(2) = -SCALE1
YFA(3) = SCALE1
YFA(4) = SCALE1
CALL GCRSG(1500)
CALL GSFAIS(1)
CALL GSFACI (4)
IF (OUT .NE. 2) THEN
CALL GFA(4,XFA,YFA)
ENDIF
CALL GCLSG

```

```

    TPOSX = . }12
    TPOSY = . }
    TPOSY1 = . }
    CALL GSCHXP (1.25)
    CALL GSCHH (.03)
    CALL GSTXFP(1,2)
    CALL GSTXAL(3,3)
    CALL GSTXCI (0)
    ```

CALL GSPLCI (1)
CALL GSLN(1)
CALL GSMK (2)
CALL GSPMCI (1)
CALL GCRSG(2000)
IF (OUT .NE. 2) THEN
CALL GPL \((360, \mathrm{X}, \mathrm{Y})\)
\(\operatorname{CALL} \operatorname{GPM}(1,0 ., 0\).
ENDIF
CALL GCLSG
K谷

IF (OUT .NE. 1) THEN
IF (OUT .EQ. 3) THEN
CALL GSELNT (3)
ELSEIF (OUT .EQ. 2) THEN
CALL GSELNT (7)
ENDIF
\(\operatorname{XFA}(1)=0\).
\(X F A(2)=1\).
\(X F A(3)=1\).
\(X F A(4)=0\).
\(\mathrm{YFA}(1)=0\).
\(\operatorname{YFA}(2)=0\).
\(Y F A(3)=1\).
\(\operatorname{YFA}(4)=1\).
为

\(\operatorname{XAXIS}(1)=0.1\)
\(\operatorname{XAXIS}(2)=0.1\)
\(\operatorname{XAXIS}(3)=0.9\)
\(\operatorname{XAXIS}(4)=0.1\)
XAXIS (5) \(=0.1\)
\(\operatorname{YAXIS}(1)=0.1\)
YAXIS (2) \(=0.5\)
YAXIS (3) \(=0.5\)
YAXIS (4) \(=0.5\)
\(\operatorname{YAXIS}(5)=0.9\)
CALL GCRSG (3000)
CALL GSFAIS (1)
CALL GSFACI (2)
CALL GFA (4,XFA,YFA)
```

    XS(1) = 0.1
    YS(1) = 0.5
    DO 800 IS=1,360
    XS (IS+1) = 0.1 + 0.8*IS/360.
    YS (IS+1) = (0.5 + 0.4*NORMS(IS))
    800 CONTINUE
CALL GSPLCI(1)
CALL GPL(5,XAXIS,YAXIS)
CALL GSPLCI(0)
CALL GPL(361,XS,YS)
CALL GTX (TPOSX,TPOSY,DIS1)
CALL GTX (TPOSX,TPOSY1,DIS2)
CALL GTX (.08,.46,'0')
CALL GTX (. 35,.46,'90')
CALL GTX (. 55,.46,'180')
CALL GTX (.75,.46,'270')
CALL GTX (.95,.46,'360')
CALL $\operatorname{GSTXAL}(2,3)$
CALL GTX (.5,.06, 'DISPLACEMENT')
CALL $\operatorname{GSTXAL}(3,3)$
CALL GCLSG
IF (OUT .EQ. 3) THEN CALL GSELNT (4)
ELSEIF (OUT .EQ. 2) THEN
CALL GSELNT (8)
ENDIF
CALL GCRSG (4000)
CALL GSFAIS (1)
CALL GSFACI (6)
CALL GFA (4, XFA, YFA)
$X V(1)=0.1$
$Y V(1)=0.5$
DO 900 IV $=1,360$
$\mathrm{XV}(\mathrm{IV}+1)=0.1+0.8 * \mathrm{IV} / 360$.
$Y V(I V+1)=(0.5+0.4 \% \mathrm{NORMV}(I V))$
900 CONTINUE
CALL GSPLCI (1)
CALL GPL (5,XAXIS,YAXIS)
CALL GSPLCI (0)
CALL GPL ( $361, \mathrm{XV}, \mathrm{YV}$ )
CALL GTX (TPOSX, TPOSY,VEL1)
CALL GTX (TPOSX,TPOSY1,VEL2)

```

CALL GTX (.08,.46, \({ }^{\prime} 0^{\prime}\) )
CALL GTX (. \(35, .46,{ }^{\prime} 90^{\prime}\) )
CALL GTX (.55,.46,'180')
CALL GTX (.75,.46,'270')
CALL GTX (. \(\left.95, .46,1360^{\prime}\right)\)
\(\operatorname{CALL} \operatorname{GSTXAL}(2,3)\)
CALL GTX (.5,.06,'VELOCITY')
CALL \(\operatorname{GSTXAL}(3,3)\)
CALL GCLSG
IF (OUT .EQ. 3) THEN
CALL GSELNT (5)
ELSEIF (OUT .EQ. 2) THEN
CALL GSELNT (9)
ENDIF
CALL GCRSG(5000)
CALL GSFAIS (1)
CALL GSFACI (2)
CALL GFA ( \(4, \mathrm{XFA}, \mathrm{YFA}\) )
\(X A(1)=0.1\)
\(\mathrm{YA}(1)=0.5\)
DO 1000 IA \(=1,360\)
\(X A(I A+1)=0.1+0.8 \div I A / 360\).
\(\mathrm{YA}(I A+1)=(0.5+0.4 \div \operatorname{NORMA}(I A))\)
1000 CONTINUE
CALL GSPLCI (1)
CALL GPL (5, XAXIS, YAXIS)
CALL GSPLCI (0)
CALL GPL (361,XA,YA)
CALL GTX (TPOSX,TPOSY,ACC1)
CALL GTX (TPOSX,TPOSY1,ACC2)
CALL GTX (.08, .46, '0')
CALL GTX (. \(35, .46, ' 90^{\prime}\) )
CALL GTX (.55,.46,'180')
CALL GTX \(\left(.75, .46,{ }^{\prime} 270^{\prime}\right)\)
CALL GTX (.95,.46,'360')
CALL \(\operatorname{GSTXAL}(2,3)\)
CALL GTX (. \(5, .06\), 'ACCELERATION')
CALL \(\operatorname{GSTXAL}(3,3)\)
CALL GCLSG
IF (OUT .EQ. 3) THEN
CALL GSELNT (6)
ELSEIF (OUT .EQ. 2) THEN
```

    CALL GSELNT (10)
    ENDIF
    CALL GCRSG(6000)
    CALL GSFAIS(1)
    CALL GSFACI (6)
    CALL GFA(4,XFA,YFA)
    XJ(1) = 0.1
    YJ(1) = 0.5
    DO 1100 IJ=1,360
    XJ(IJ+1) = 0.1 + 0.8*IJ/360.
    YJ(IJ+1) = (0.5 + 0.4*NORMJ(IJ))
    1100 CONTINUE
CALL GSPLCI(1)
CALL GPL(5,XAXIS,YAXIS)
CALL GSPLCI(0)
CALL GPL(361,XJ,YJ)
CALL GTX (TPOSX,TPOSY,JER1)
CALL GTX (TPOSX,TPOSY1,JER2)
CALL GTX (.08,.46,'0')
CALL GTX (.35,.46,'90')
CALL GTX (.55,.46,'180')
CALL GTX (.75,.46,'270')
CALL GTX (.95,.46,'360')
CALL GSTXAL}(2,3
CALL GTX (.5,.06,'JERK')
CALL GSTXAL}(3,3
CALL GCLSG
ENDIF

```
```

WRITE (6,*) ' Do you want to animate this cam? (Y/N)'

```
WRITE (6,*) ' Do you want to animate this cam? (Y/N)'
READ (5,10)CHAR
READ (5,10)CHAR
    IF (CHAR .EQ. 'Y') THEN
1200 WRITE (6,*)' Enter the degree step you want'
READ(5,*)ITE
WRITE(6,*) ' Transformations being performed.'
WRITE (6,*) ' Please wait.'
CALL GSELNT (2)
CALL GSPLCI(1)
CALL GSMK(2)
CALL GSPMCI(1)
APHI = 0.
NSTEP = 360/ITE
```

```
    DO 1300 IANI = 1,NSTEP
    APHI = APHI + ITE*PI/180.
    CALL GCRSG(IANI)
    CALL GSVIS(IANI,0)
    CALL GPL(360,X,Y)
    CALL GPM(1,0.,0.)
    CALL GEVTM(0.,0.,0.,0.,APHI,1.,1.,0,XFRM)
    CALL GSSGT(IANI,XFRM)
    CALL GCLSG
1300 CONTINUE
    WRITE (6,%) ', Transformations Completed.'
1400 WRITE (6,%) ' How many rotations do you want to see?'
    WRITE (6,*) ' MIN: 1 MAX: 30'
    READ(5,*)IRO
    IF (IRO .LT. 1 .OR. IRO .GT. 30) THEN
        WRITE (6,*)' ERROR!! Input not valid. Reenter value'
        GOTO 1300
    ENDIF
1 4 0 0 ~ C A L L ~ G S V I S ( 2 0 0 0 , 0 )
    DO 1700 IAX = 1, IRO
        CALL GSVIS(2000,1)
        CALL GSVIS(2000,0)
        DO 1500 IANI = 1,NSTEP
            CALL GSVIS(IANI,1)
            CALL GSVIS(IANI,0)
        CONTINUE
1700 CONTINUE
    CALL GSVIS(1500,1)
    CALL GSVIS(2000,1)
```

```
WRITE(6,*) ' Animation Complete.'
```

WRITE(6,*) ' Animation Complete.'
WRITE (6,*) ' Do you want to see it again? (Y/N)'
WRITE (6,*) ' Do you want to see it again? (Y/N)'
READ (5,10)CHAR
READ (5,10)CHAR
IF (CHAR .EQ. 'Y') THEN
IF (CHAR .EQ. 'Y') THEN
WRITE(6,*) ' Do you want to change degree step.(Y/N)?'
WRITE(6,*) ' Do you want to change degree step.(Y/N)?'
READ(5,10)CHAR
READ(5,10)CHAR
IF (CHAR .EQ. 'Y') GOTO 1200
IF (CHAR .EQ. 'Y') GOTO 1200
WRITE (6,*) ' Do you want to change number of rotations? (Y/N)'
WRITE (6,*) ' Do you want to change number of rotations? (Y/N)'
READ (5,10)CHAR
READ (5,10)CHAR
IF (CHAR .EQ. 'Y') GOTO 1400*
IF (CHAR .EQ. 'Y') GOTO 1400*
GOTO 1400

```
GOTO 1400
```

ENDIF

ENDIF

CALL GDAWK (1)
CALL GCLWK (1)
CALL GCLKS
CLOSE (20)
RETURN
END




## SUBROUTINE CADAM

COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF COMMON/COUNT/ I, N, M, B, PI, C, R, L COMMON/ANGLE/ BETA, THETA, GAMMA
COMMON/MTN/ E(25), H(20), FROM(20), UPTO (20)
COMMON/CRV/ S (360), V(360), A(360), J(360)
COMMON/COOD/ ALPHA(360), X(361), Y(361), Z(361)
COMMON/DISP/ OUT, MONI
INTEGER I, N, M, C, R, W
INTEGER TYPE, OUT, DESI, MONI
INTEGER E, FROM, UPTO

REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
REAL ROLL, ECC, RA, RF
REAL S, V, A, J
REAL ALPHA, X, Y, Z, H
CHARACTER DRAWID*20, USERID*8, GROUP*4, IDVU*4, TEXT*12
CHARACTER ID1*4, ID2*4, ID3*4
REAL XVCTR(3), YVCTR(3), ZVCTR(3)
REAL XVTTR(3), YVTTR(3), ZVTTR(3)
REAL XYZ1 (3), XYZ2(3), XYZ3(3), XYZ4(3)
REAL $\operatorname{D} 1(3,46), D(3,46), \operatorname{VCAD}(3,46), \operatorname{SCAD}(46)$
REAL T1(46), T2(46), T3(46)

DATA DRAWID/'ANICAM ARG'/
DATA USERID/'CAM '/, GROUP/'MECH'/
DATA XVCTR/1.,0.,0./,YVCTR/0.,0.,1./, ZVCTR/0.,-1.,0./
DATA XVTTR/0.866,-0.5,0.5/,YVTTR/0.5,0.866,0./
DATA ZVTTR/0.5,-0.5,-0.866/
DATA IDVU/'IDIS'/
DATA ID1/'LIN1'/, ID2/'LIN2'/, ID3/'LIN3'/
DATA NOUT/6/
$N=46$
$M=3$
$\mathrm{K}=0$
IOPT $=1$
$100 \operatorname{WRITE}(6, *)^{\prime}$
Enter shaft radius'
READ ( $5, *$ ) SHAFT
IF (SHAFT .LT. 0) THEN
WRITE $\left(6,{ }^{*}\right)^{\prime} \quad$ ERROR!! RADIUS MUST BE POSITIVE' GOTO 100
ENDIF

CALL CADST(DRAWID,USERID,GROUP)

CALL BEGVU(IOPT,IDVU,XYZ,XVTTR,YVTTR,ZVTTR,*1000)
EXTR $=0$.
$Z(1)=0$.
$Z(2)=$ WIDTH
DO 400 IZ $=1,2$
DO 200 ICOOR = 1, 45

```
ANGL = 8*ICOOR*PI/180.
D1(1,ICOOR) = SHAFT:COS (ANGL)
D1(3,ICOOR) = SHAFT*SIN(ANGL)
D1(2,ICOOR) = Z(IZ)
```

200 CONTINUE

```
D1 (1,46) = D1 (1, 1)
D1(3,46) = D1 (3,1)
D1(2,46) = Z(IZ)
```

K
CALL SPLINE ( $\mathrm{N}, \mathrm{M}, \mathrm{K}, \mathrm{D} 1, \mathrm{VCAD}, \mathrm{SCAD}, \mathrm{T} 1, \mathrm{~T} 2, \mathrm{~T} 3, * 1001$ )
(

DO 300 ICOOR $=1,45$
ICORD $=8 \div$ ICOOR
$D(1, I C O O R)=X(I C O R D)$
$D(3, I C O O R)=Y(I C O R D)$
$D(2, I C O O R)=Z(I Z)$
300 CONTINUE
$D(1,46)=X(8)$
$D(3,46)=Y(8)$
$D(2,46)=Z(I Z)$
M
CALL SPLINE (N,M,K,D,VCAD , SCAD $, T 1, T 2, T 3, * 1001)$

## 400 CONTINUE

K

```
DO 500 I = 1,45
XYZ1(1) = D(1,I)
XYZ1(3) = D(3,I)
XYZ1(2) = 0.
XYZ2(1) = D(1,I)
XYZ2(3) = D(3,I)
XYZ2(2) = WIDTH
XYZ3(1) = D1(1,I)
XYZ3(3) = D1(3,I)
XYZ3(2) = 0.
```

```
XYZ4(1) = D1(1,I)
XYZ4(3) = D1(3,I)
XYZ4(2) = WIDTH
```

CALL LINE3D（XYZ1，XYZ2，＊1004）
CALL LINE3D（XYZ3，XYZ4，＊1004）
CALL LINE3D（XYZ1，XYZ3，＊1004）
CALL LINE3D（XYZ2，XYZ4，＊1004）
500 CONTINUE

IOPT＝2
NOGOOD＝0
CALL CADFIL（IOPT，NOGOOD，IDUMMY）
IF（NOGOOD．NE．1）THEN
WRITE（NOUT，＊）＇ERROR IN CADFIL＇
ENDIF
GOTO 2000

```
1000 WRITE(NOUT,*)'ERROR EXISTS IN BEGVU'
1001 WRITE (NOUT,*)'ERROR EXISTS IN SPLINE'
1002 WRITE(NOUT,*)'ERROR EXISTS IN NAME'
1003 WRITE(NOUT,*)'ERROR EXISTS IN SRFRUL'
1004 WRITE(NOUT,*)'ERROR EXISTS IN LINE3D'
2000 RETURN
END
```


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$\therefore \quad$ THIS SUBROUTINE DISPLAYS THE OUTPUT PARAMETERS OF THE CAM TO A \％THE SCREEN IN FORM OF 3－D GRAPHICS．IT GENERATES THE GEOMETRY
＊：FILE（FILE．CAM）FOR DISPLAY，USING MOVIE．BYU．IF REQUIRED，IT
＊：CREATES A COMMAND FILE（FILE．COM）TO BE USED IN CONJUNCTION
\＃WITH THE ANIMATE SOFTWARE DEVELOPED BY A．MYKLEBUST．
＊THIS ROUTINE IS CALLED ONLY IF THE DISPLAY SYSTEM CHOSEN BY＊＊
स THE USER AT THE BEGINING OF THE PROGRAM IS MOVIE．

## SUBROUTINE MOVIE

COMMON/CAM1/ TYPE, BASE, RPM, WIDTH, DESI, ROLL, ECC, RA, RF
COMMON/COUNT/ I, N, M, B, PI, C, R, L
COMMON/ANGLE/ BETA, THETA, GAMMA
COMMON/MTN/ E(25), H(20), FROM(20), UPTO (20)
COMMON/CRV/ S 360 ), V(360), A(360), J(360)
COMMON/COOD/ ALPHA (360), X(361), Y(361), Z(361)
COMMON/DISP/ OUT, MONI
INTEGER I, N, M, C, R, W
INTEGER TYPE, OUT, DESI, MONI
INTEGER E, FROM, UPTO
REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
REAL ROLL, ECC, RA, RF
REAL S, V, A, J
REAL ALPHA, X, Y, Z, H
REAL XI $(3,150)$, XS (361), YS $(361)$
INTEGER IP(600), $\operatorname{NPL}(2,2)$
CHARACTER*8 FILE
CHARACTER*1 CHAR
$\mathrm{IJ}=1$
$N P=1$
$\mathrm{NJ}=148$
NPT= 144
NCON $=576$
$\mathrm{NPL}(1,1)=1$
$\operatorname{NPL}(2,1)=144$
$25 \operatorname{WRITE}(6, *)^{\prime} \quad$ Enter shaft radius'
READ (5,*)SHAFT
IF (SHAFT .LT. 0) THEN
WRITE $(6, *)$
ERROR!! RADIUS MUST BE POSITIVE'
GOTO 25
ENDIF
*
DO $50 \mathrm{I}=1,360$
ANGL $=\mathrm{I} \div \mathrm{PI} / 180$.
XS (I) $=$ SHAFT*COS (ANGL)
YS $(I)=$ SHAFT*SIN(ANGL)

50 CONTINUE

```
XS(361) = XS(1)
YS(361) = YS(1)
X(361) = X(1)
Y(361) = Y(1)
```

. N to be used for the representation

$$
\begin{aligned}
& \text { DO } 100 \mathrm{I}=1,361,10 \\
& \mathrm{X} 1(1, \mathrm{IJ})=\mathrm{X}(\mathrm{I}) \\
& \mathrm{X} 1(2, \mathrm{IJ})=Y(I) \\
& \mathrm{X} 1(3, \mathrm{IJ})=0 . \\
& \mathrm{IJ}=\mathrm{IJ}+1 \\
& 100 \text { CONTINUE }
\end{aligned}
$$

DO $200 \mathrm{I}=1,361,10$
$X 1(1, I J)=X S(I)$
$\mathrm{X1}(2, \mathrm{IJ})=\mathrm{YS}(\mathrm{I})$
$X 1(3, I J)=0$.
$I J=I J+1$
200 CONTINUE
DO $300 \mathrm{I}=1,74$
$\mathrm{X} 1(1, \mathrm{I}+74)=\mathrm{X} 1(1, \mathrm{I})$
$\mathrm{X} 1(2, I+74)=X 1(2, I)$
$\mathrm{X} 1(3, \mathrm{I}+74)=\mathrm{WIDTH}$
300 CONTINUE
DO $400 \mathrm{I}=1,36$
$\operatorname{IP}(4 * I-3)=I$
$\operatorname{IP}(4 \div I-2)=I+1$
$\operatorname{IP}(4 \div \mathrm{I}-1)=\mathrm{I}+38$
$\operatorname{IP}(4 * I)=-(I+37)$
400 CONTINUE
DO $500 \mathrm{I}=37,72$
$\operatorname{IP}(4 * I-3)=I-36$
$\operatorname{IP}(4 \div I-2)=I+38$
$\operatorname{IP}(4 * I-1)=I+39$
$\operatorname{IP}(4 * I)=-(I-35)$
500 CONTINUE
DO $600 \mathrm{I}=73,108$
$\operatorname{IP}(4 * I-3)=I-35$
$I P(4 \div I-2)=I+39$
$I P(4 \div I-1)=I+40$
$I P(4 \div I)=-(I-34)$
600 CONTINUE
DO $700 \mathrm{I}=109,144$
$\mathrm{IP}(4 \times \mathrm{I}-3)=\mathrm{I}-34$
$\operatorname{IP}(4 \div I-2)=I-33$
$\operatorname{IP}(4 * I-1)=I+4$
$\operatorname{IP}(4 * I)=-(I+3)$
700 CONTINUE
高

REWIND (8)
WRITE (8, 800) NP, NJ, NPT, NCON
$\operatorname{WRITE}(8,800)((\operatorname{NPL}(I, I J), I=1,2), I J=1, N P)$
WRITE $(8,900)((X 1(I, I J), I=1,3), I J=1, N J)$
WRITE $(8,800)(\mathrm{IP}(\mathrm{I}), \mathrm{I}=1, \mathrm{NCON})$
800 FORMAT (16I5)
900 FORMAT (6E12.5)
WRITE $(6, *)^{\prime}$ Do you want to animate this cam?'
$\operatorname{READ}(5,1000)$ CHAR
IF (CHAR .EQ. 'N') GOTO 2000
\%
REWIND (10)
1000 FORMAT(A1)
WRITE $(6, *)$ 'GIVE NAME OF DISK FILE'
$\operatorname{READ}(5,1100)$ FILE
1100 FORMAT(A8)
$\operatorname{WRITE}(6, *)$ 'GIVE STEP ANGLE'
READ (5,*) STEP
NSTEP $=360 /$ STEP
WRITE (9,*)'FILE.CAM'
WRITE $(9,1200)$
1200 FORMAT (/)
$\operatorname{WRITE}(9, *)^{\prime}$ CLSG'
DZ $=360 /$ NSTEP
ZANG $=0$.
DO 1500 IJ $=1$,NSTEP
WRITE (9,*)'ROTA'

```
    WRITE (9,*)'Z'
    WRITE (9,1300)ZANG
1300 FORMAT(F10.4)
    WRITE (9,*)'OSEG'
    WRITE (9,1400)IJ
1400 FORMAT(I2)
    WRITE(9,*)'VIEW'
    WRITE (9,*)'CSEG'
    ZANG = DZ
1500 CONTINUE
    WRITE (9,*)'SAVT'
    WRITE (9,1600)FILE
1600 FORMAT(A8,/,'EXIT')
2 0 0 0 ~ R E T U R N
    END
```



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\% SUBROUTINE DOCUM 安 安
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\#: THIS SUBROUTINE SAVES THE LAST CAM DESIGNED BY THE USER. THE **
\#: DOCUMENTATION IS DONE UNDER THE FILE CAM.OUT. THIS HELPS THE \$*
** USER TO MAINTAIN RECORDS OF THE CAMS DESIGNED AND THEN SELECT **
动 THE BEST OF ALL OF THEM.
*: THIS ROUTINE IS CALLED ONLY IF THE USER WANTS THE DESIGN TO **

* BE SAVED. **
※\% +



SUBROUTINE DOCUM
COMMON／CAM1／TYPE，BASE，RPM，WIDTH，DESI，ROLL，ECC，RA，RF
COMMON／COUNT／I，N，M，B，PI，C，R，L COMMON／ANGLE／BETA，THETA，GAMMA COMMON／MTN／E（25），H（20），FROM（20），UPTO（20） COMMON／CRV／S（360），V（360），A（360），J（360） COMMON／COOD／ALPHA（360），X（361），Y（361），Z（361） COMMON／DISP／OUT，MONI

INTEGER I，N，M，C，R，W
INTEGER TYPE，OUT，DESI，MONI
INTEGER E，FROM，UPTO

REAL BETA, THETA, GAMMA, B, PI, BASE, RPM, WIDTH
REAL ROLL, ECC, RA, RF
REAL S, V, A, J
REAL ALPHA, X, Y, Z, H
*
REWIND (8)
DO $100 \mathrm{~K}=10,360,10$
WRITE ( 8,10 ) $\mathrm{K}, \mathrm{S}(\mathrm{K}), \mathrm{V}(\mathrm{K}), \mathrm{A}(\mathrm{K}), \mathrm{X}(\mathrm{K}), \mathrm{Y}(\mathrm{K})$
10 FORMAT (1X,I3,5(2X,F7.3))
100 CONTINUE

RETURN
END

# The vita has been removed from the scanned document 

